# 1:1 Interactive Architecture Prototypes TU Delft Q3 2023/24

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### LUNASCAPE



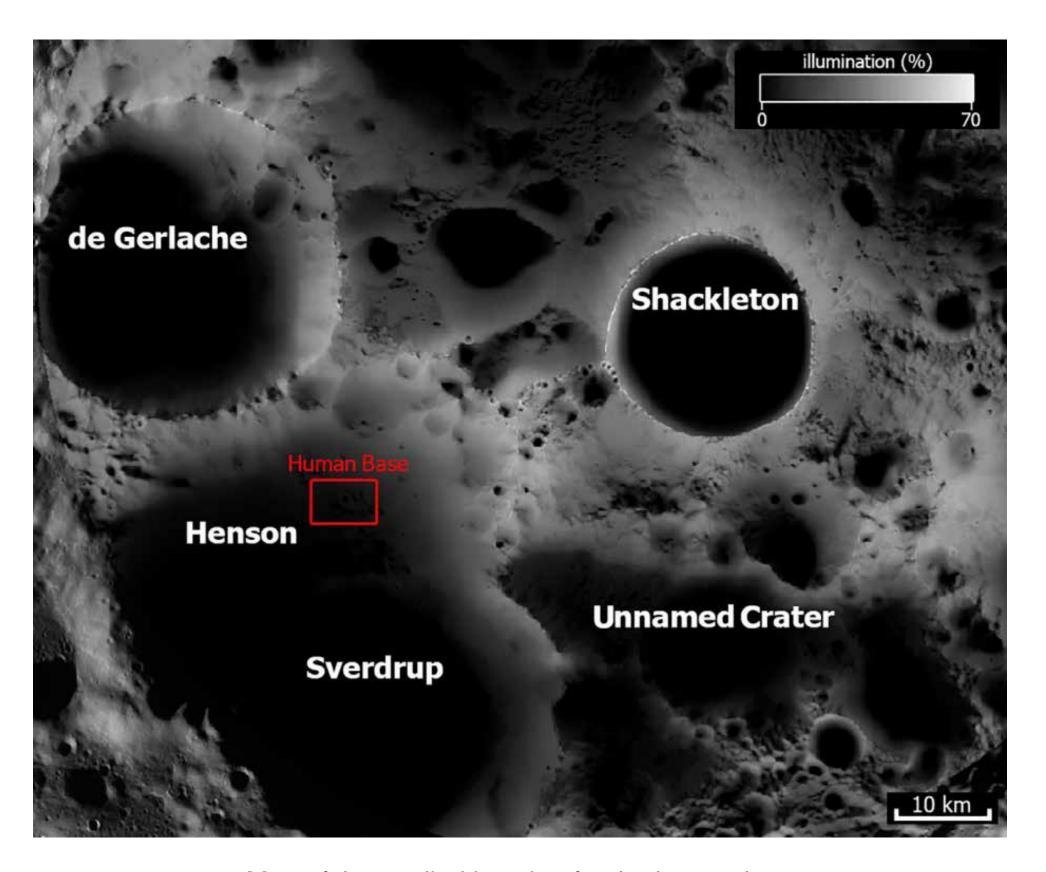
# 1 LOCATION AND SITE



### SITE ANALYSIS

### **Lunar South Sverdrup-Henson crater (location 1)**

- Approximately 5 km² of area.
- •Flat topography inside the crater.
- •Abundant water supply, nearby shaded areas have more ice and materials.
- •Parts of the crater are covered by **sunlight all day long**, which is suitable for solar power generation.
- •The terrain is **suitable to build ground antennas** for connection with the Earth.

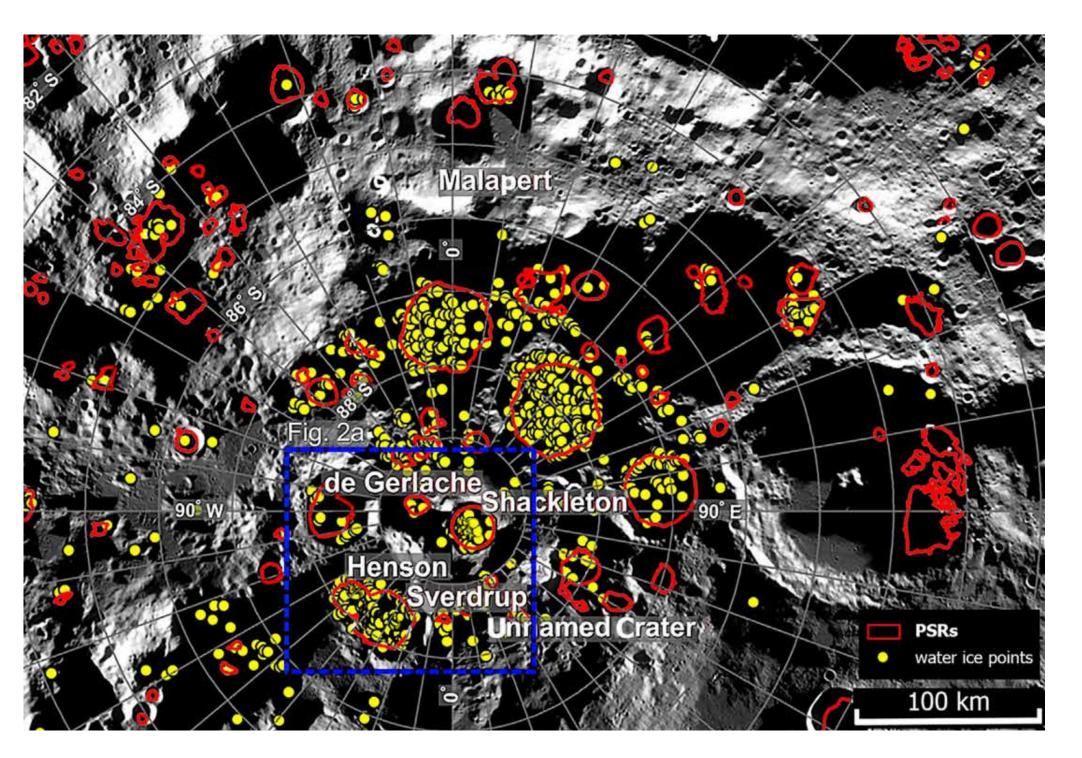


Map of the studied location for the human base



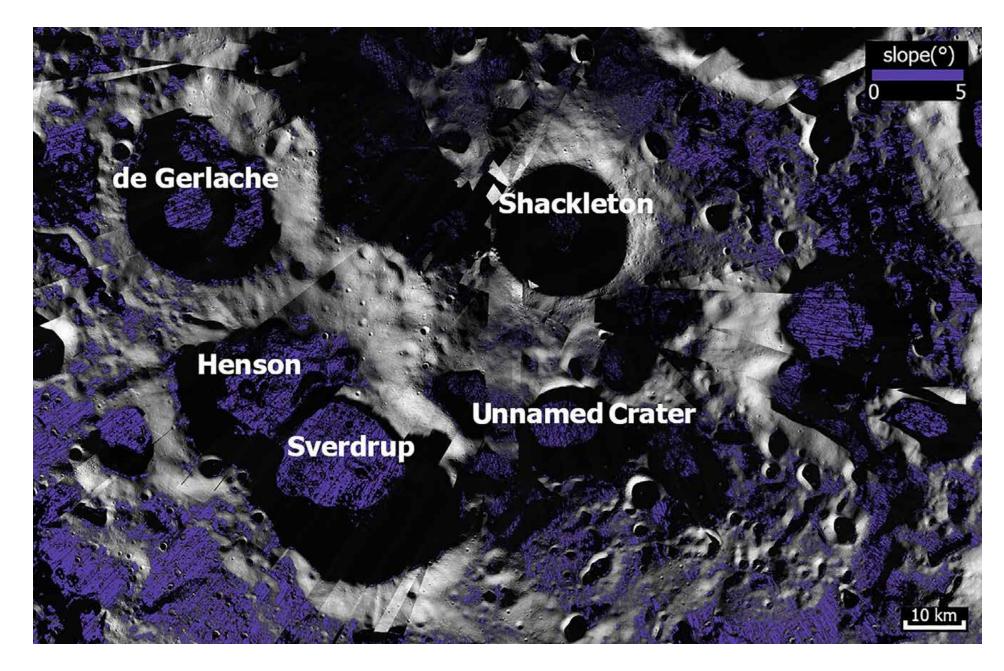
### **PSRs** and water ice points

The image shows the water ice points available in the area. As it can be seen, the reason why the Lunar South Pole is such a strong candidate for a Lunar settlement is the large concentration of these points. PSR is an acronym for permanently shadowed regions. Of course, surface water ice points are located mainly in the PSRs, as they would evaporate if they became in contact with the sun.



Map of the studied location for the human base

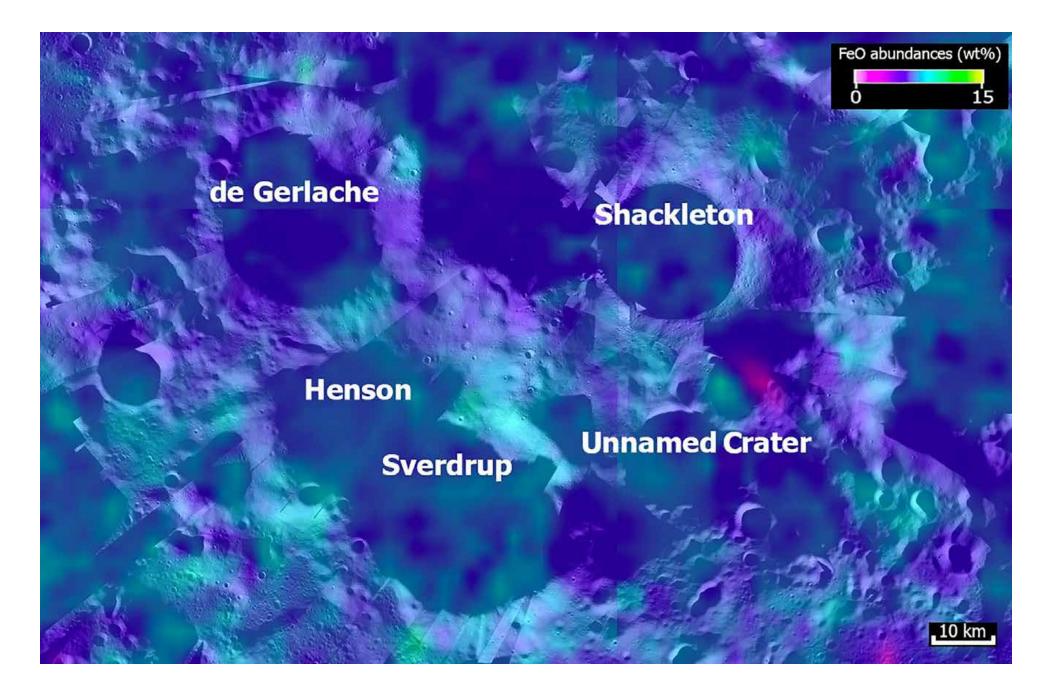




Slope of the terrain in the considered area.

### **Terrain slope**

The slope of the terrain is also very suitable. Ideally, a flat surface would be the most desirable, but moderate slope angles may still be considered relatively safe, depending on the task. For example, a slope of 7° allows spacecraft landing, and mobile surface operations are safe on an angle of up to 15°. The image shows the slopes between 0° and 5° in the area.



Map of iron oxide abundance

#### **Available mineral resources**

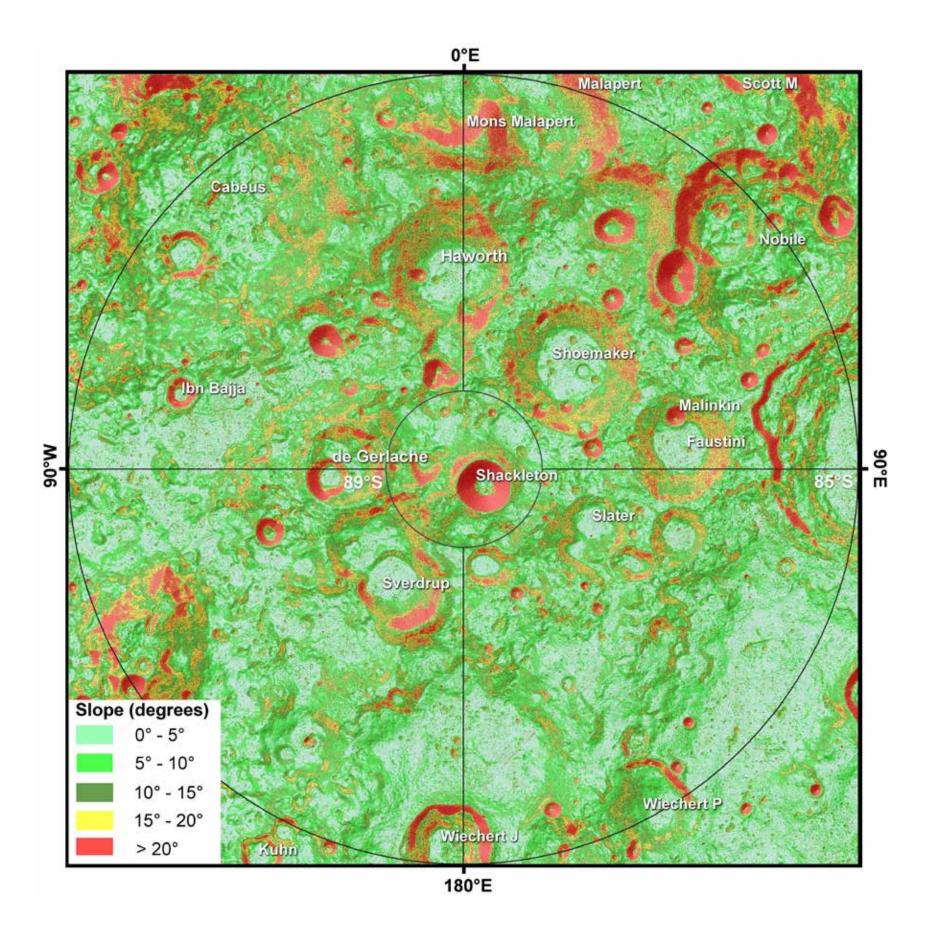
Another crucial factor involves the availability of mineral resources essential for constructing technological equipment through In-Situ Resource Utilization (ISRU). Notably, iron and titanium oxides, as well as rare earth elements, play a key role. These materials are prominently found in the Oceanus Procellarum KREEP Terrane (PKT), particularly in the eastern part of the Em4 geological unit, but recent sampling has provided insights into this region. Additionally, the area is rich in rare earth elements, with concentrations of up to 4.6 wt. % yttrium and up to 0.25 wt. % neodymium.



### **CARTOGRAPHIES**

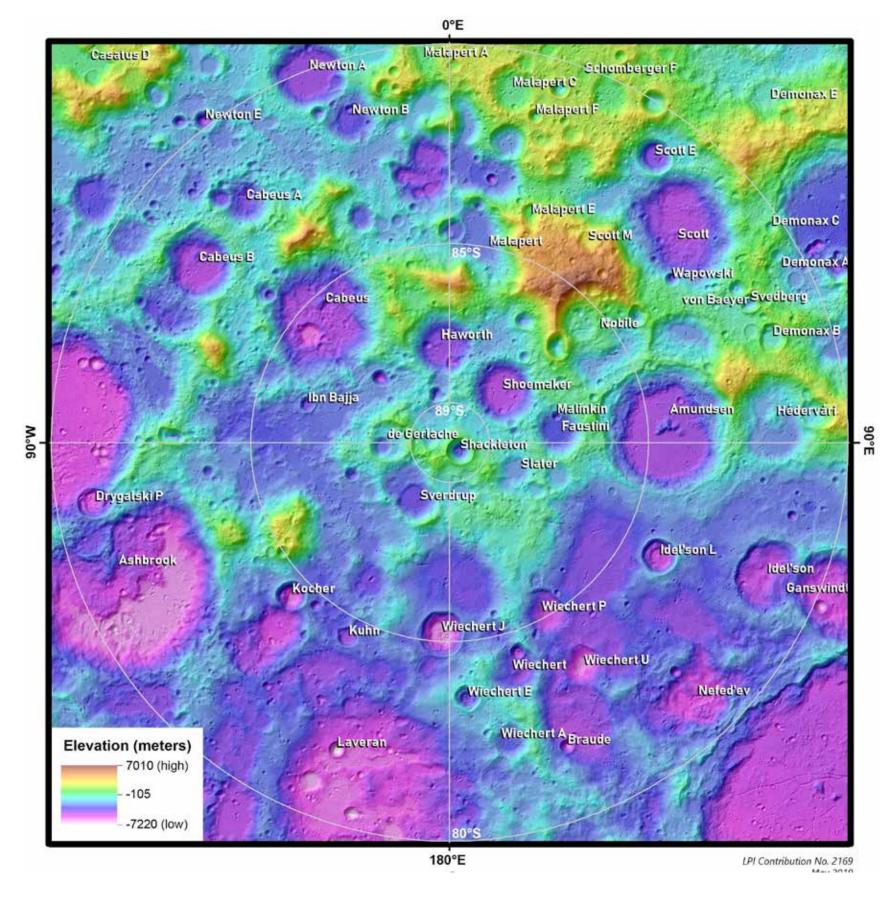
### From the Lunar South Pole Atlas

The following slides show some interesting and relevant cartographic maps developed by the **Center for Lunar Science** and **Exploration**. They showcase **high quality**, relevant site data such as elevation, slope, and near surface temperatures, among others, of the investigated area.

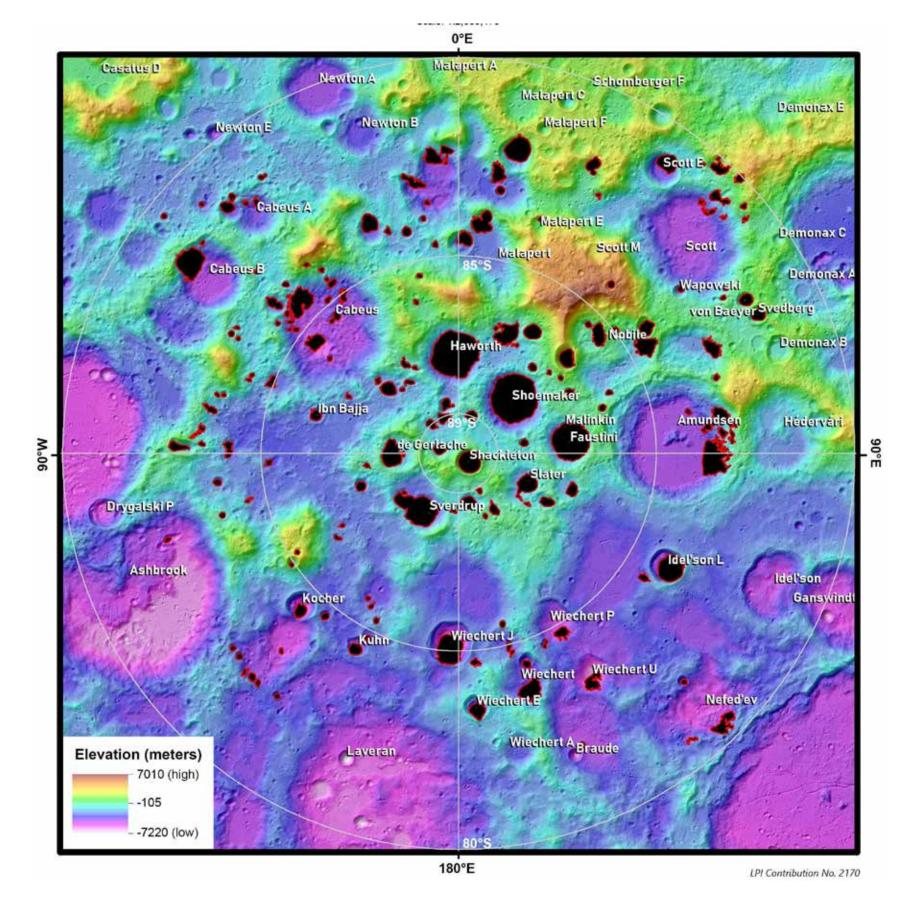


Detailed slope map, in 5 degree breaks





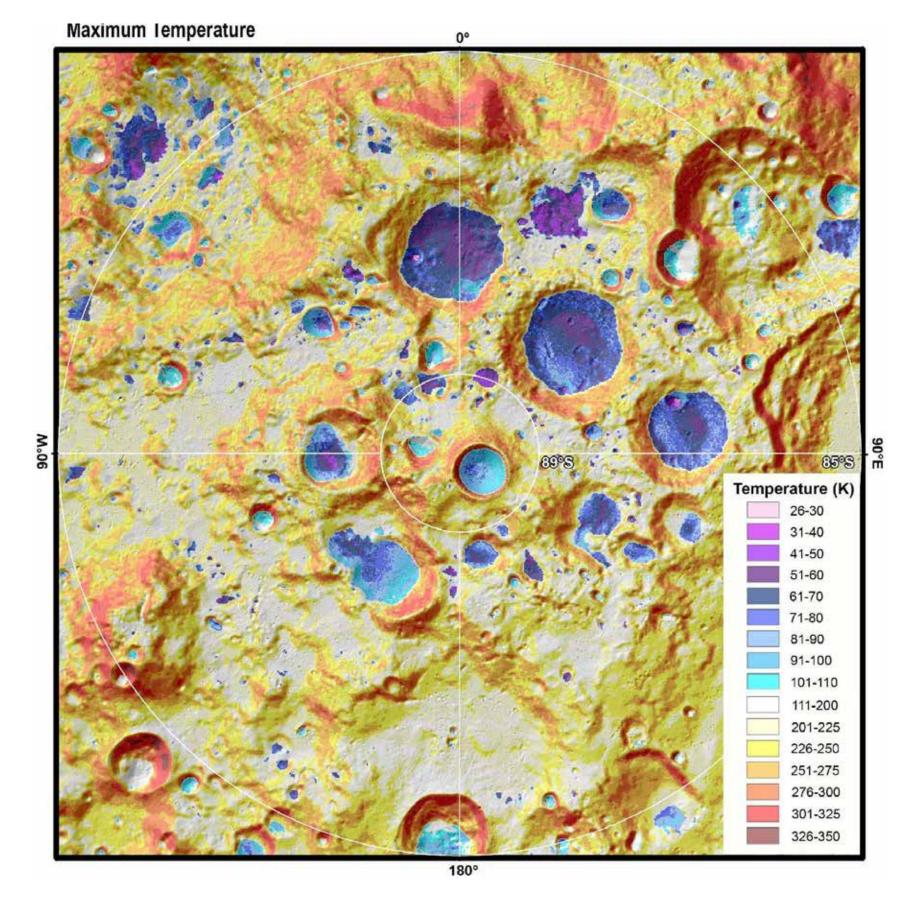
Elevation map, in meters. The height difference is **greater than 14000 meters** between the highest and lowest points.



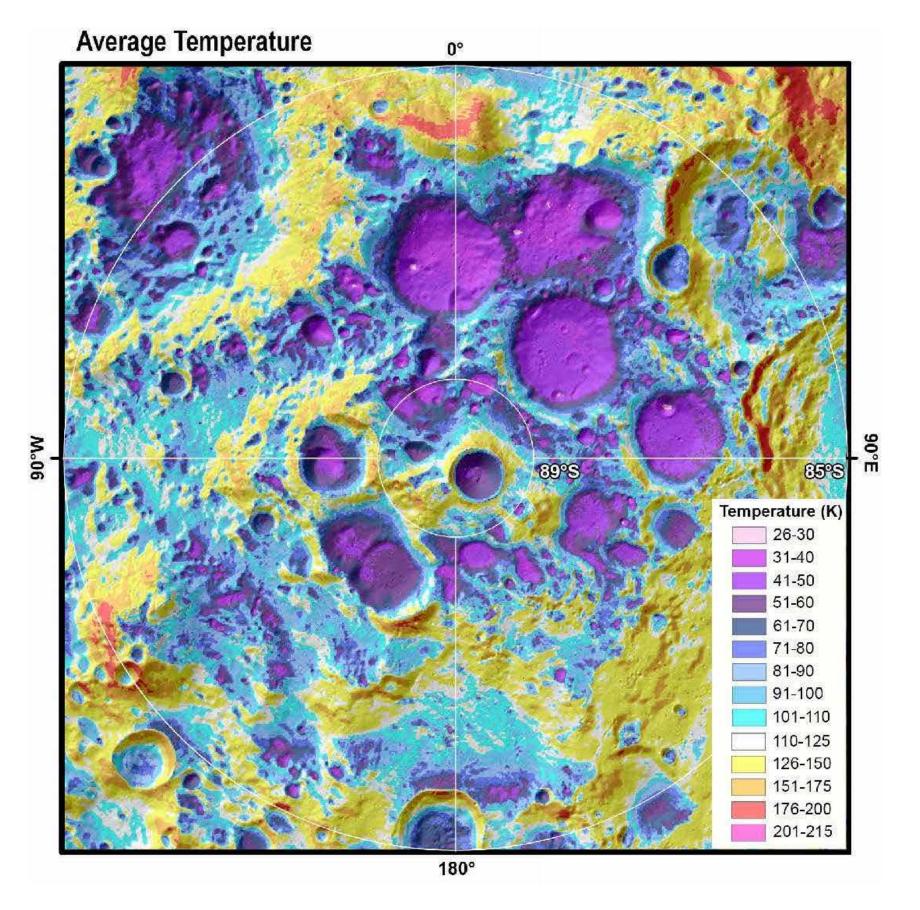
The **permanently shadowed regions (PSRs)** overlayed on the elevation map.

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Maximum temperatures map, in Kelvin. They're notably lower on craters and PSRs.



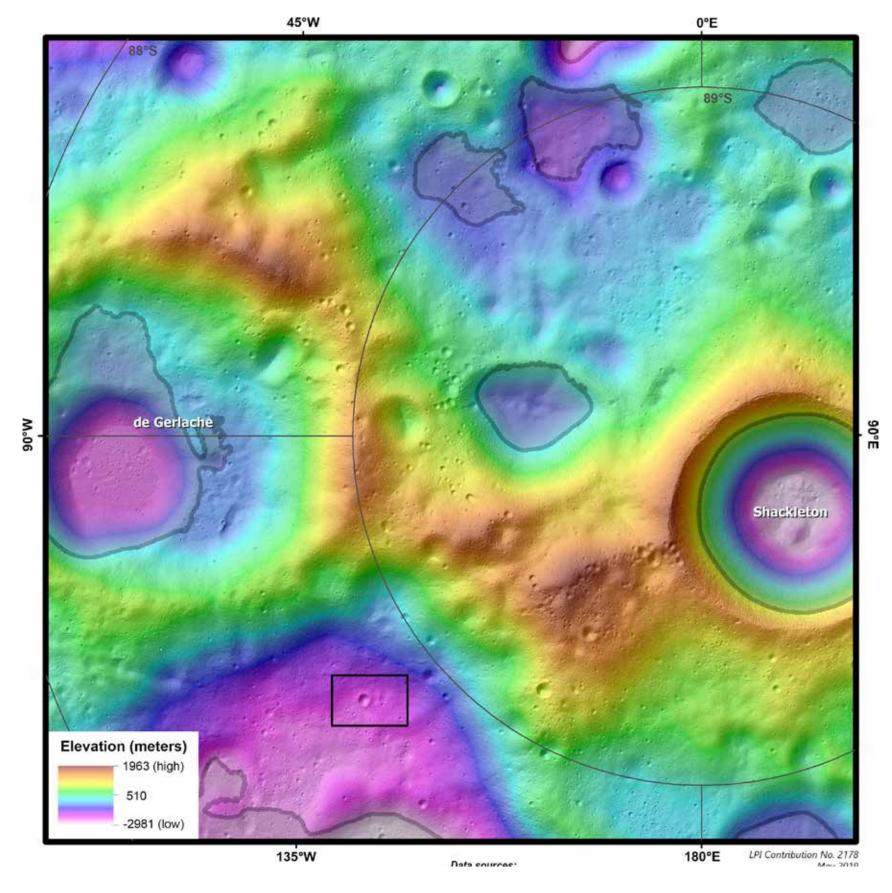
Average temperatures map, in Kelvin

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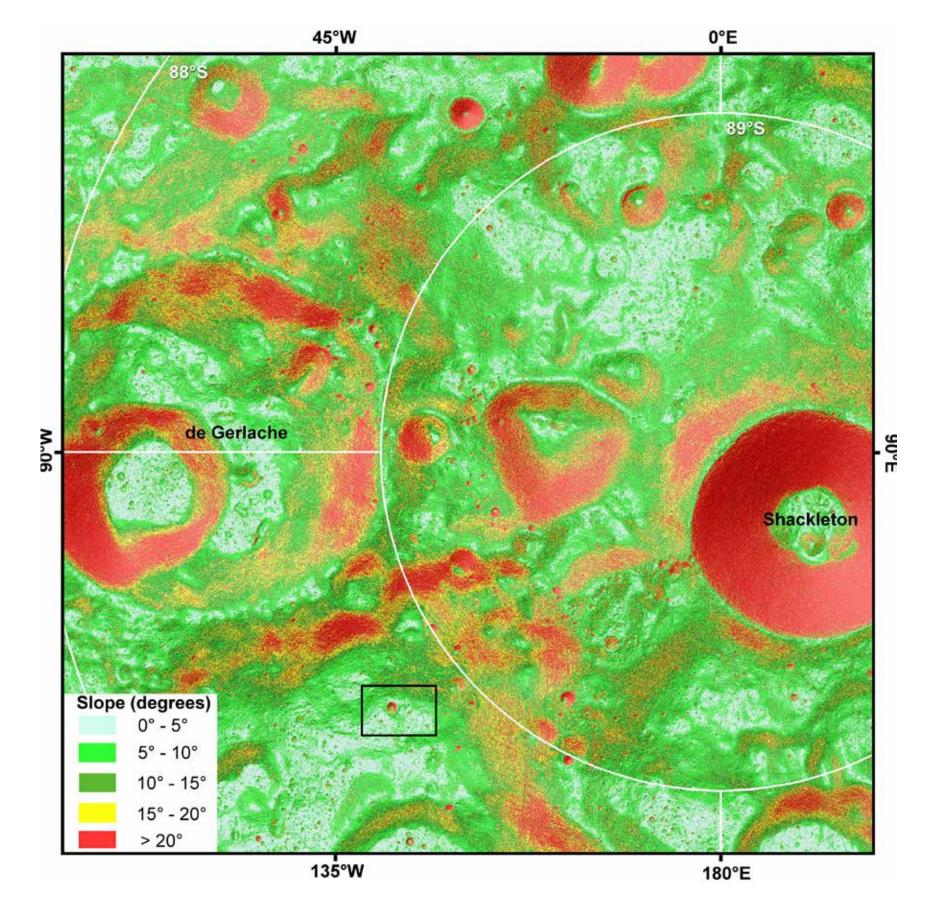




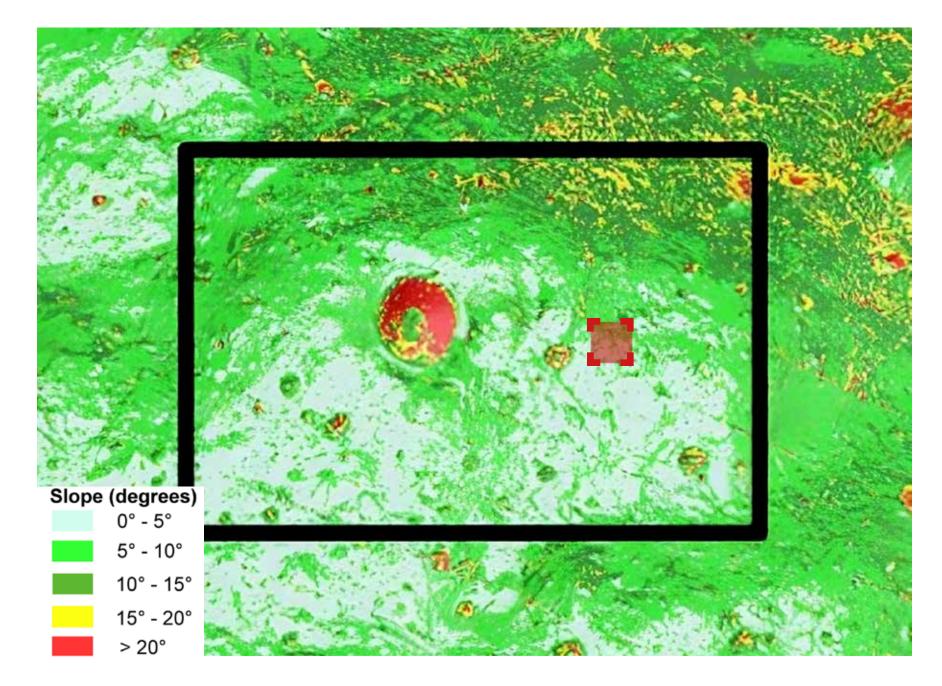
### Site-overlaid close up cartographies:



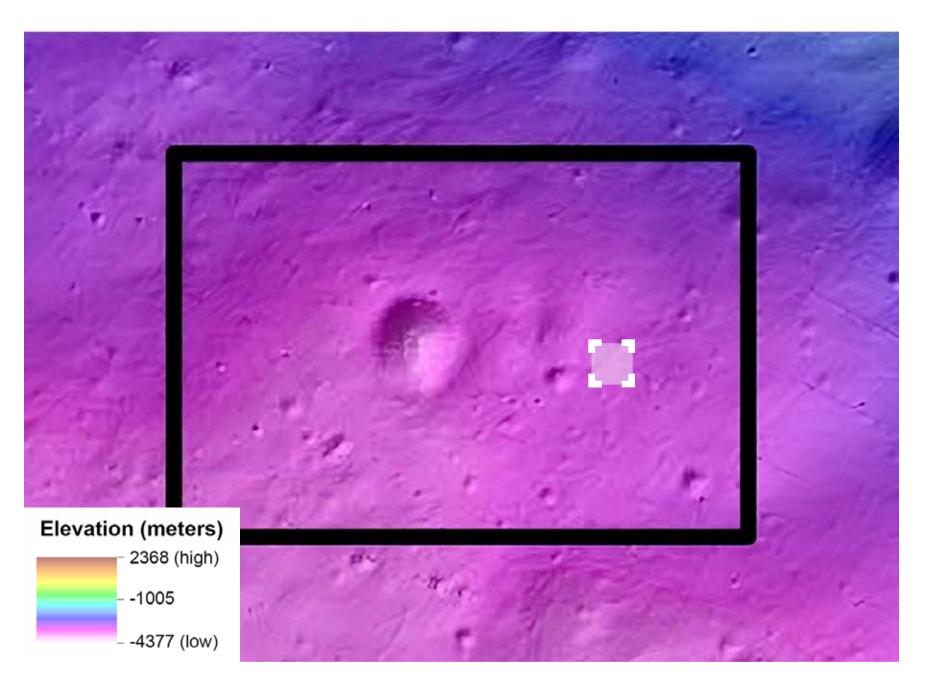
Detailed slope map, in 5 degree breaks



Detailed slope map, in 5 degree breaks



Slope map with site location



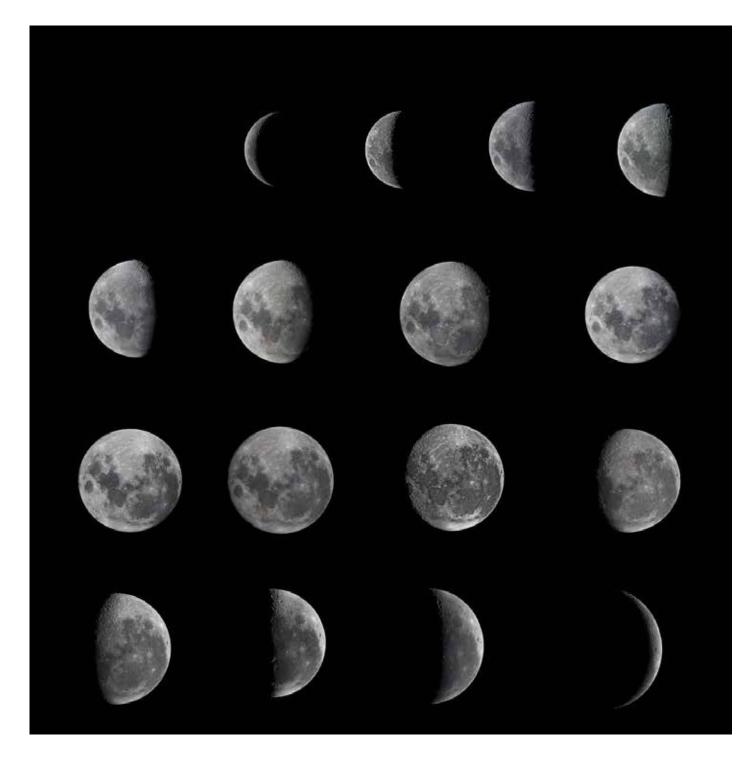
Slope map with site location





### CLIMATIC ANALYSIS

- •Extreme temperature conditions (from +127 °C to -173 °C). In the PSRs the temperature can be even lower, as temperatures of up to -246 °C have been recorded.
- •Radiation from the Sun is very intense, more than a hundred times that of Earth: the measured surface radiation in the Moon is 60  $\mu$ Sv/hr, while on Earth it usually remains below 0.2  $\mu$ Sv/hr.
- •Gravity is 1.62 m/s², one sixth of Earth's.
- •A Lunar day, that is, the time it takes the Moon to complete on its axis one synodic rotation, takes 29.5 Earth days. That means approximately 350 hours of continuous Sun exposure and heating and another 350 hours of darkness and cold.



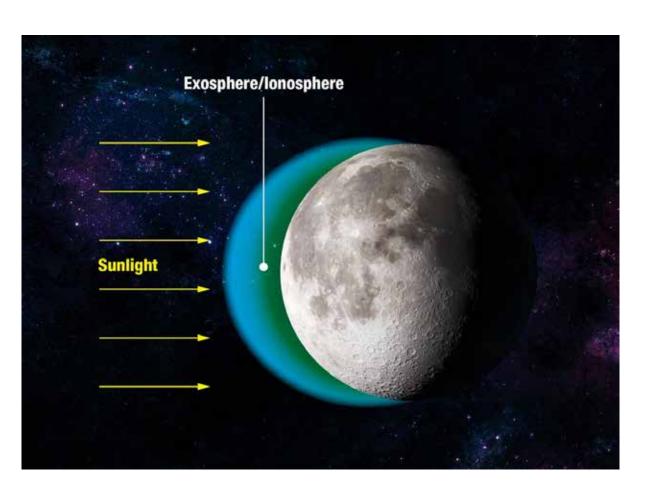
Lunar day cycle



• Micrometeoroids, and sometimes larger space objects, impact the lunar surface on a regular basis. This dry shower of debris shuffles materials in the Moon's exterior layers, exposing fresh material in a process known as impact gardening.

• There is a very thin type of atmosphere on the Moon, known as an **exosphere**, which contains **helium**, **argon**, **neon**, **ammonia**, **methane** and **carbon** dioxide. The exosphere is **not breathable**, and **in the cold lunar night it falls to the ground**.





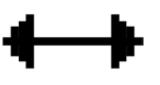
# 2 PROGRAMS AND NEEDS

















**BERTHING** 

**RECREATION** 

WORKSPACE

**EXERCISE** 

**HYGEINE** 

STORAGE

SAFETY BUNKER

Docking station for vehicles

exits

Emergency

Sleeping pods (private)

Dining space (communal)

Lounging spaces (communal)

Multifunctioning laboratories

Hydroponics

Multifunctioning exercise spaces

Bathing and bathroom spaces (communal)

Multi-use storage zone

Self-contained underground bunker in case of emergency

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DOCKING
STATION

EMERGENCY
EXIT ZONE

VOLUME	TIME SPENT USING SPACE	ABOVE OR BELOW GROUND	INTERNAL RISK RATING	EXIT POINTS BASED ON INTERNAL RISK
60 m <sup>3</sup>	Variable time	Above ground	2	Docking Station: 2+ exits
	Docking proce-	Easy access for		
	dures can last	shipping deploy-		Emergency Exit
	from several min- utes to a few	ments		Zones 2+ exits
	hours.			ZT CXII3
	15 mins - 3 hrs			



SLEEPING PODS (PRIVATE)

DINING SPACE (COMMUNAL)

LOUNGING (COMMUNAL)

VOLUME	TIME SPENT USING SPACE	ABOVE OR BELOW GROUND	INTERNAL RISK RATING	EXIT POINTS BASED ON INTERNAL RISK
Sleeping Pods - 45 m³ total	Sleeping - 8 hrs	Below ground	Sleeping - 1	Sleeping Pods 1 exit
9 m³/ person	Dining - 30 mins to	High security	Dining - 2	
	1 hr per meal, max			Dining Space
Dining - 20 m <sup>3</sup>	3 hrs per day		Lounging - 1	2 exits
Lounging - 20 m <sup>3</sup>	Lounging - varia- ble time, estimate 3 hrs			Lounging Spaces (Communal) 2 exits





HYDRAPONICS GREENHOUSE

VOLUME	TIME SPENT USING SPACE	ABOVE OR BELOW GROUND	INTERNAL RISK RATING	EXIT POINTS BASED ON INTERNAL RISK
Laboratories	Variable time	Sub-level - anal-	Laboritories - 3	Laboritories
100 m <sup>3</sup>		ysis of upper		2 exits
	Laboratories - 5 hrs	surface and con-	Hydroponics - 2	
Hydroponics		trolled labs be-		Hydroponics
Greenhouse 80 m <sup>3</sup>	Hydroponics - 5 hrs	neath the surface		2 exits
	In conjunction with	Controlled space -		
	Lunar Surface Activities - 5 hrs	high-security		



MULTI-PURPOSE EXERCISE SPACES (COMMUNAL)

VOLUME	TIME SPENT USING SPACE	ABOVE OR BELOW GROUND	INTERNAL RISK RATING	EXIT POINTS BASED ON INTERNAL RISK
40 m³	1-2 hrs a day	Above ground	1	2 exits
		Interaction with sunlight		



BATHROOM
AND
BATHING SPACES
(COMMUNAL)

VOLUME	TIME SPENT USING SPACE	ABOVE OR BELOW GROUND	INTERNAL RISK RATING	EXIT POINTS BASED ON INTERNAL RISK
30 m³	1 hr a day	Below ground	1	Bathroom
		Controlled space		1 exit
				Bathing
				1 exit



MULTI-PURPOSE STORAGE FACILITY

VOLUME	TIME SPENT USING SPACE	ABOVE OR BELOW GROUND	INTERNAL RISK RATING	EXIT POINTS BASED ON INTERNAL RISK
40 m <sup>3</sup>	Variable time	Below ground	2	2 exits
	Estimate 2 hrs	Highly controlled space		
	Access as needed	High security		



UNDERGROUND EMERGENCY BUNKER SELF-CONTAINED

VOLUME	TIME SPENT USING SPACE	ABOVE OR BELOW GROUND	INTERNAL RISK RATING	EXIT POINTS BASED ON INTERNAL RISK
100 m <sup>3</sup>	Variable time	Below ground	2	2 exits
	In case of emergency	High security		
	Estimate of 8 hrs			

# 3 ASSEMBLY AND CONSTRUCTION



### LUNAR CHALLENGES

#### **LUNAR SOIL IS DANGEROUS**

Potential of acute and/or chronic multiorgan toxicities No direct wall contact, no regolith can be carried inside

#### **ASSEMBLY MUST BE AIRTIGHT**

1 bar pressure and breathable atmosphere

#### **OPTIONS**

Binder, Spray/Glaze, Membrane, Tiles, Tube that can be extended

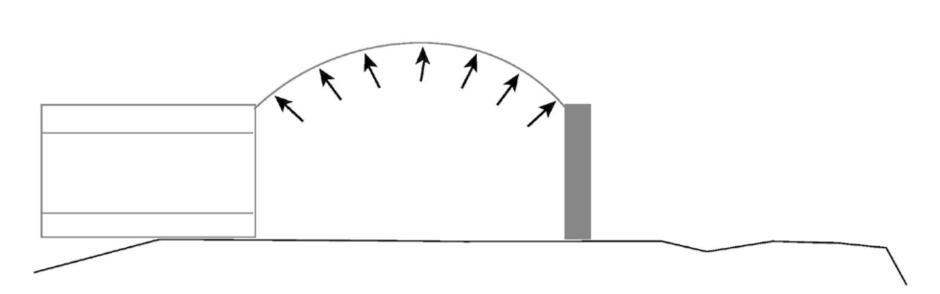
Pohlen, M., Carroll, D., Prisk, G.K. et al. Overview of lunar dust toxicity risk. npj Microgravity 8, 55 (2022). https://doi.org/10.1038/s41526-022-00244-1



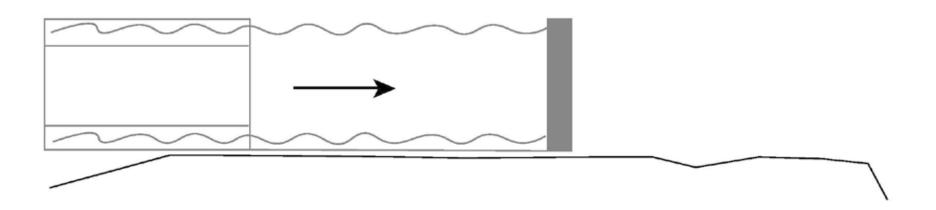
# **Assembly Process**



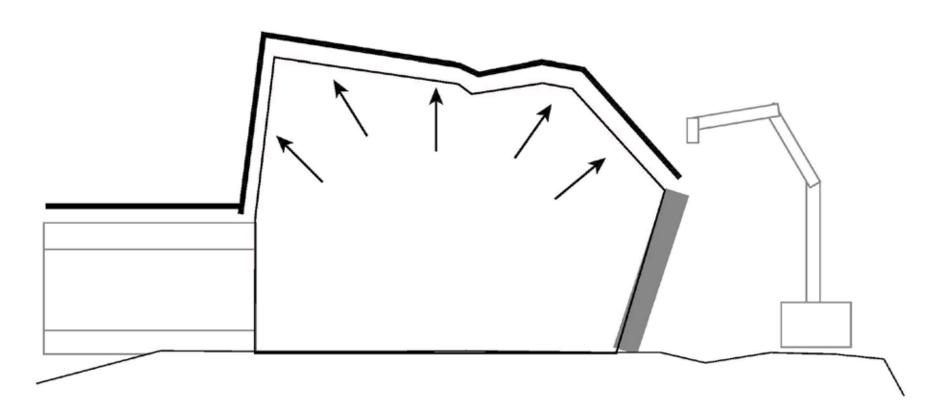
Step 0: Airlock



**Step 2: Membrane inflation** 

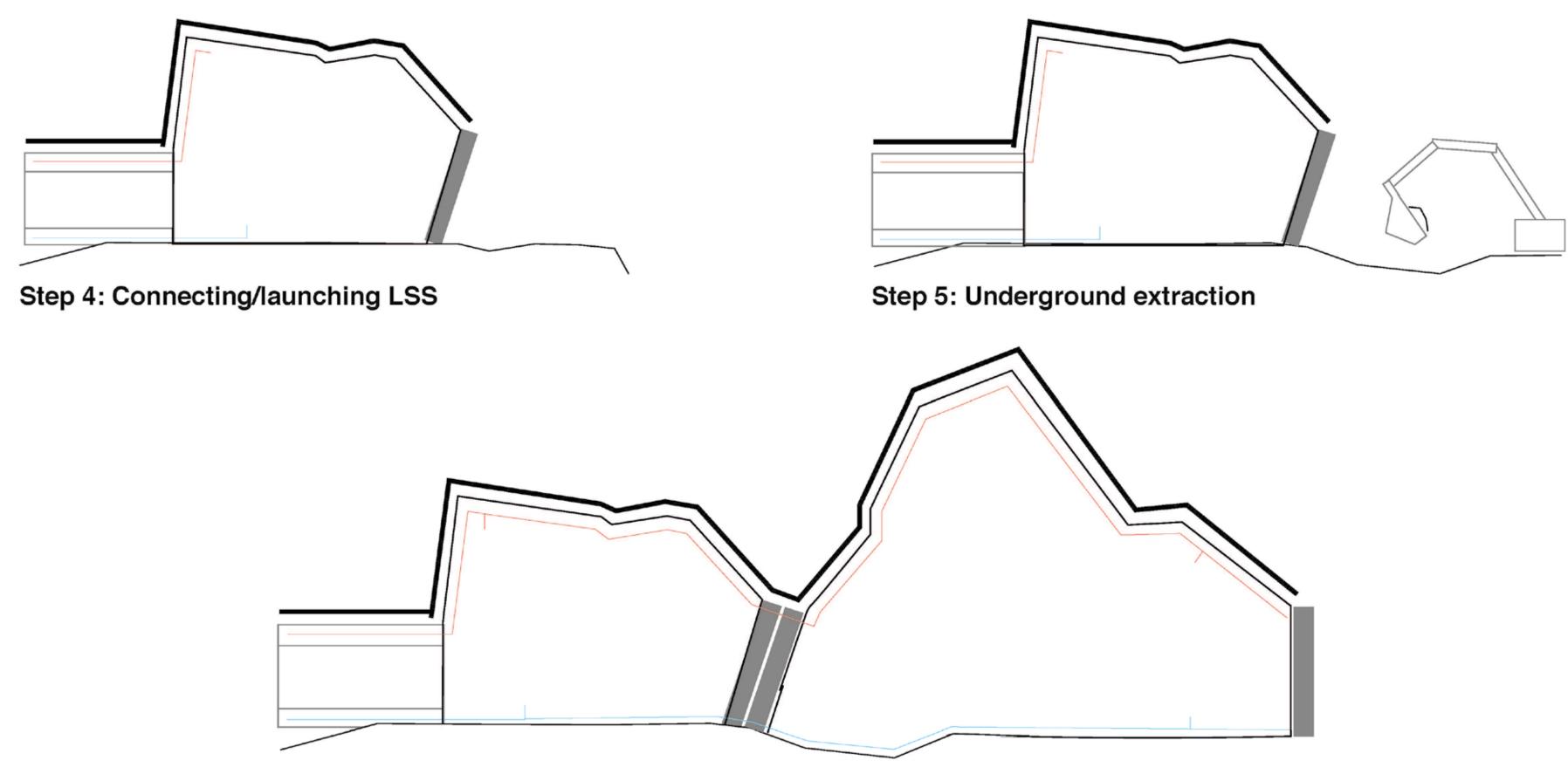


**Step 1: Opening process** 



Step 3: Voronoi-shell printing and connecting it to membrane

# **Assembly Process**



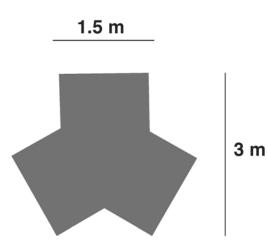
Step 6: Repeating the process for the next module



## Components - Airlock

### **Geometry and Dimensions:**

Isotoxal-star-form with three entrances to ensure modular connectivity, safe emergency routes and pollution-free entering/leaving the module

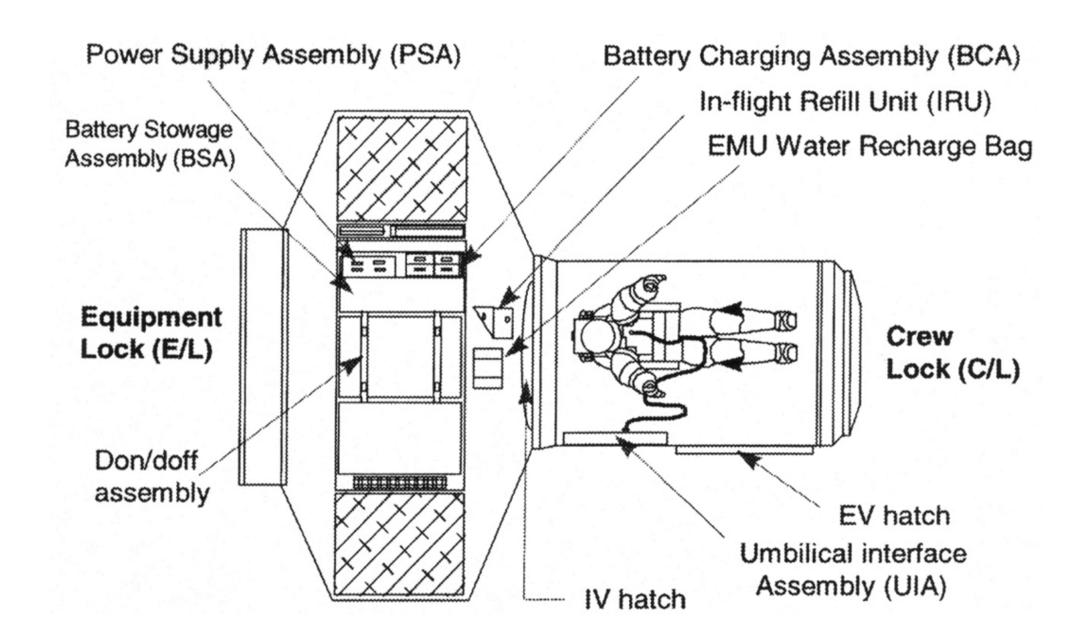


#### **Entrance:**

Voroni- module entrance: always the same shape as airlock

### **Functionality:**

holds the LSS
provides electricity access
seals habitat
prevents habitat from regolith pollution



https://www.lpi.usra.edu/lunar/artemis/Mary-2018-EVA%20Airlocks-And-Alternative-Ingress-Egress-EVA-EXP-0031.pdf

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### Components - Membrane

### **Adaptable Geometry:**

Flexible fabric that can fit modules of x m3 (large membrane) or y m3 (small membrane)

### **Integrated Sensor Array:**

Oxygen, Carbondioxide, Particles, Smoke, Temperature

### **Radiation Protection and Airtightness:**

While the 3D print is still wet, the mebrane can be attached to the shell with the help of the soft spikes that can be pressed into the regolith layer



https://arstechnica.com/science/2019/08/one-could-fly-to-mars-in-this-spacious-habitat-and-not-go-crazy/



# **Components - Overview**

Pressure Chamber,

**Energy Storage** 

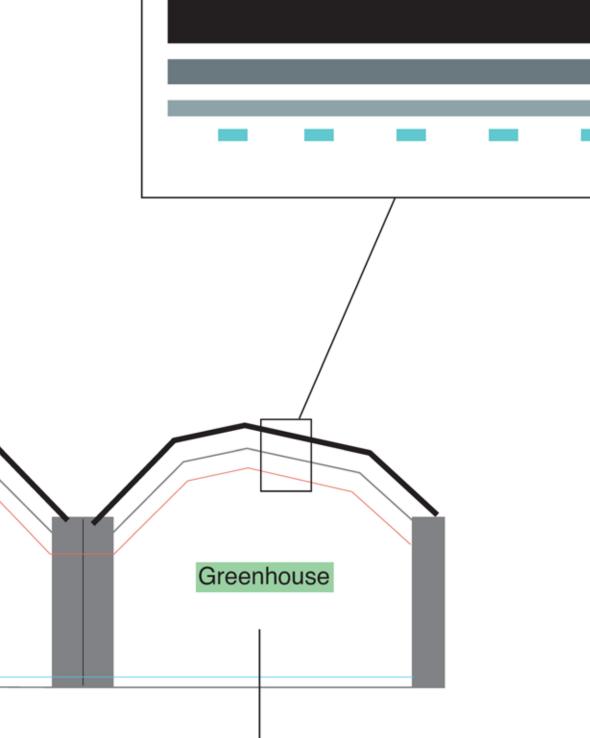
Water

Air Supply/Ventilation

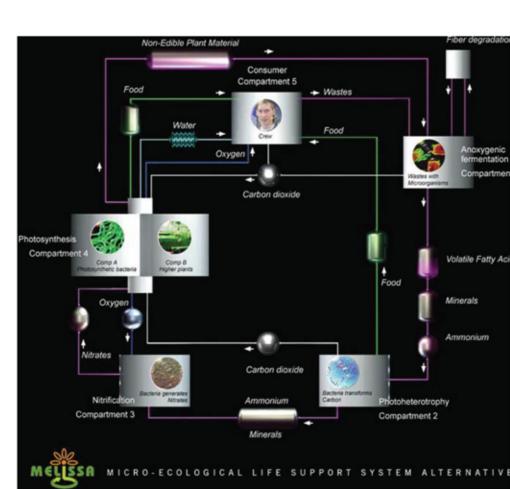
**Suit Storage** 

Cables: Flexible Pipes:

Base: 3 cables Base: 2 pipes Add-on: 2 cables Add-on: 1 pipe



Food



**3D Printed Regolith** 

with solid spikes

**Sensor Array** 

Thermal Shield (Mylar)

Air Barrier (Vectron)

Radiation Shield (Polyethylene)

https://link.springer.com/chapter/10.1007/ 978-3-030-52859-1\_3

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PV





# 4 \_ ENERGY AND MATERIALITY

### **Radiation Protection**

**Exposure:** Galactic cosmic radiation (GCR), Electromagnetic radiation (EM), Charged Particles (Protons, Electrons), Solar Particle Events (SPE) + Secondary Radiation (neutrons)

https://link.springer.com/content/pdf/10.1007/978-3-319-14541-9\_179.pdf

#### **Material Choice:**

Radiation Protection of 3d printed regolith (Rhizome)? Geopolymer Binder Lunamer?

https://www.sciencedirect.com/science/article/pii/S0273117715004019?casa\_token=7WhQc2vwIhwAAAAA:NRXUCd7Kc8sTs4IKwNj6Riw-WHTAOGvcNLF8M-QIrrLvIkbLl4B7iw5M1bdZYVbHET15xUZ1B2g

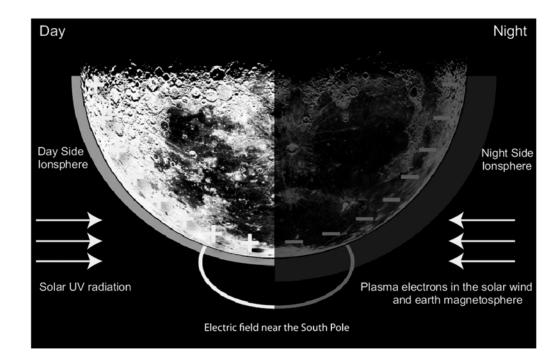
1) Recycled high-density polyethylene plastic (r-HDPE) reinforced with ilmenite mineral (IIm) Flexible "fabric", not a structural material

https://pubs.rsc.org/en/content/articlehtml/2023/ra/d3ra03757f

2) Hydrogenated BNNT (nanotubes constructed of carbon, boron, nitrogen, hydrogen) Can be structural

https://www.nasa.gov/general/radiation-shielding-materials-containing-hydrogen-boron-and-nitrogen-systematic-computational-and-experimental-study/

Wall Thickness
Habitat Position (surface or subsurface, which room where)
Window Number and Position (direct or no direct radiation exposure)



https://www.researchgate.net/figure/Dayand-night-difference-on-the-Lunar-South-Polar-location-solar-UV-radiation-causes-the\_ fig2\_230853777



### **Energy**

https://www.sciencedirect.com/science/article/pii/S0094576521002289?casa\_token=Ga-QD7l0HaoAAAA:nEM-Sux46FxqeKvr2\_Tp2508QrPT4Db4axNo1hEhVqsNK0BTzpwGjh\_Bv9f-3a9bIRu3y22AlWw

**Demand:** depends mainly on its inventory and usage profile, along with the round trip efficiency of the applied energy storage system (ESS)

#### **Generation:**

Electricity: PV (efficiency, light exposure on site)

https://www.sciencedirect.com/science/article/pii/S030626192100266X?casa\_token=0MxkHRTDgaQAAAAA:HL-tO-ni-lCin-\_4osWvUwUlW1x0UjQ\_Z-NP8-CBVq608W1HOF-McjSFoUSzC098FQs3lSAhGPQ#f0010

Heat: Internal heat gains?

Food: Melissa LSS and Greenhouse

#### Storage:

Lithium-Ion Batteries Regenerative Fuel Cells Lithium-Sulphur Batteries

- -> Find out energy demand (KW) for core unit (x astronauts, y rooms)
- -> Include growth of PV, batteries, greenhouse, LSS etc. in script

#### **Considerations:**

- total power demand of the base;
- daily power load management strategy;
- type of applied power source;
- type of solar array structure;
- type of energy storage or energy buffering system;
- the base location (selenographic latitude);
- solar illumination conditions

# 5 PARAMETRIC INPUTS AND DESIGN

### Timeline of the design process:

		Room Concepts		Safety Requ	uirements			
		GH Scripting				Robotic		3D Printing
Research	Concept Ge	neration			Concept	Detailing 		
Lunar Conditions	Hybrid (surface a	nd subsurface)	Life Support	System Integration	Radiation	Protection	Airlock Connection	
Terrain Analysis	Parametric Ro	om Scaling	Construction	n/Assembly Process	Inflatab	le Membrane	Energy Generation	
Voronoi Research								
Astronauts' Needs								

### INPUT FACTORS

#### **SIZE**

Slider 0-100 metre cubed

### TIME SPENT (SECURITY)

Wall Thickness 2h x 40cm Toggle Underground

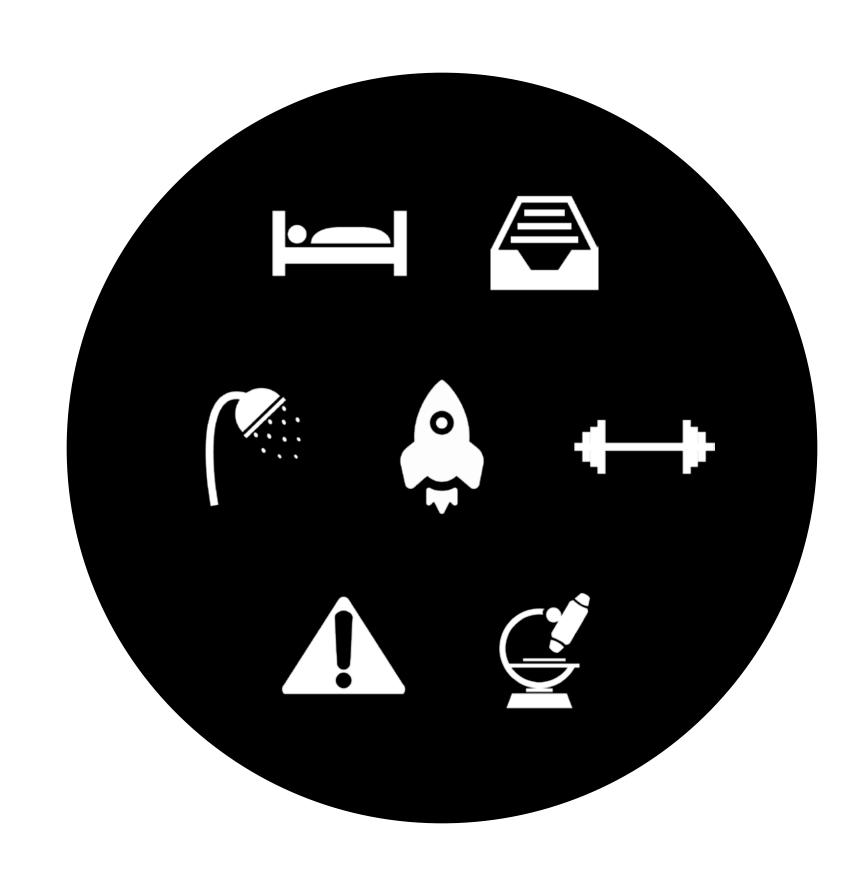
#### **INTERNAL RISK**

Risk Levels
Determine Distance from the Centre

### **ROOM CONNECTIONS**

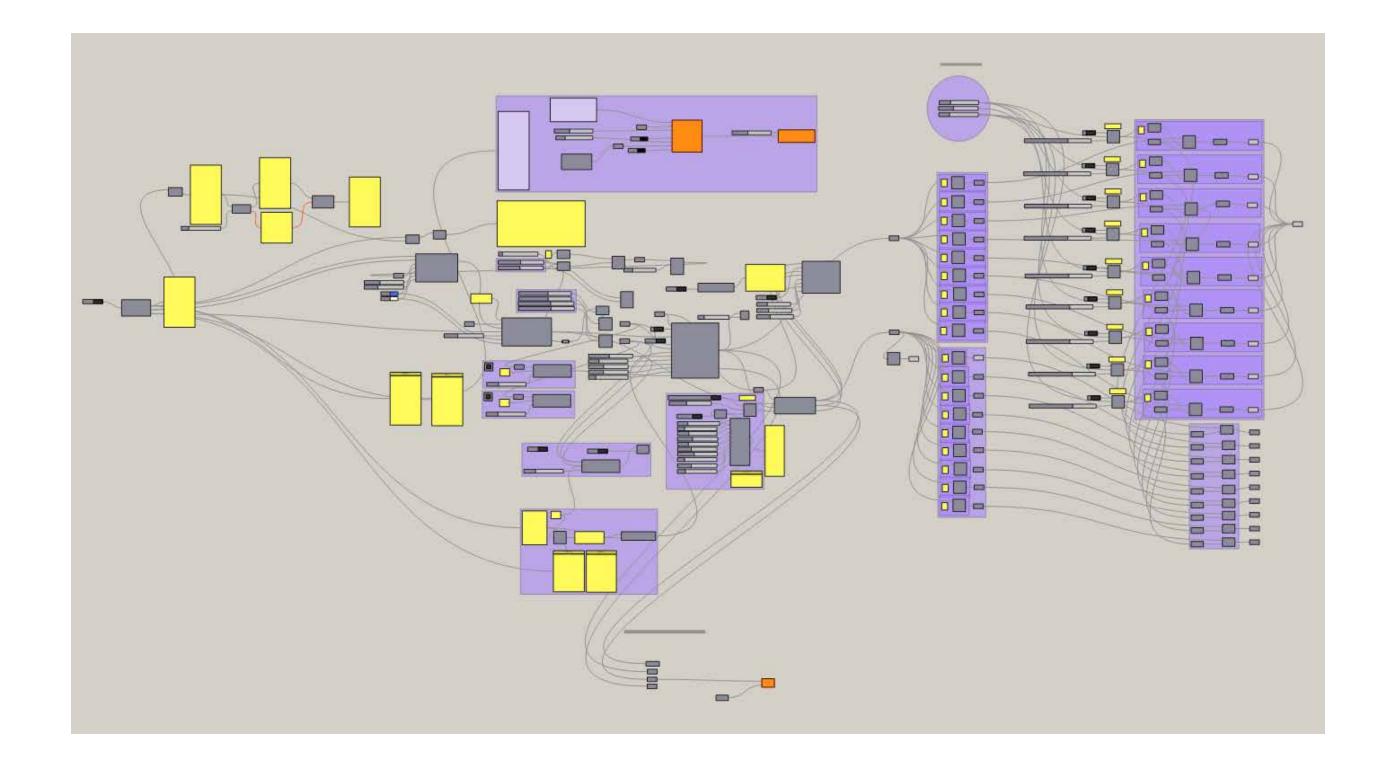
#### DOORS/OPENINGS

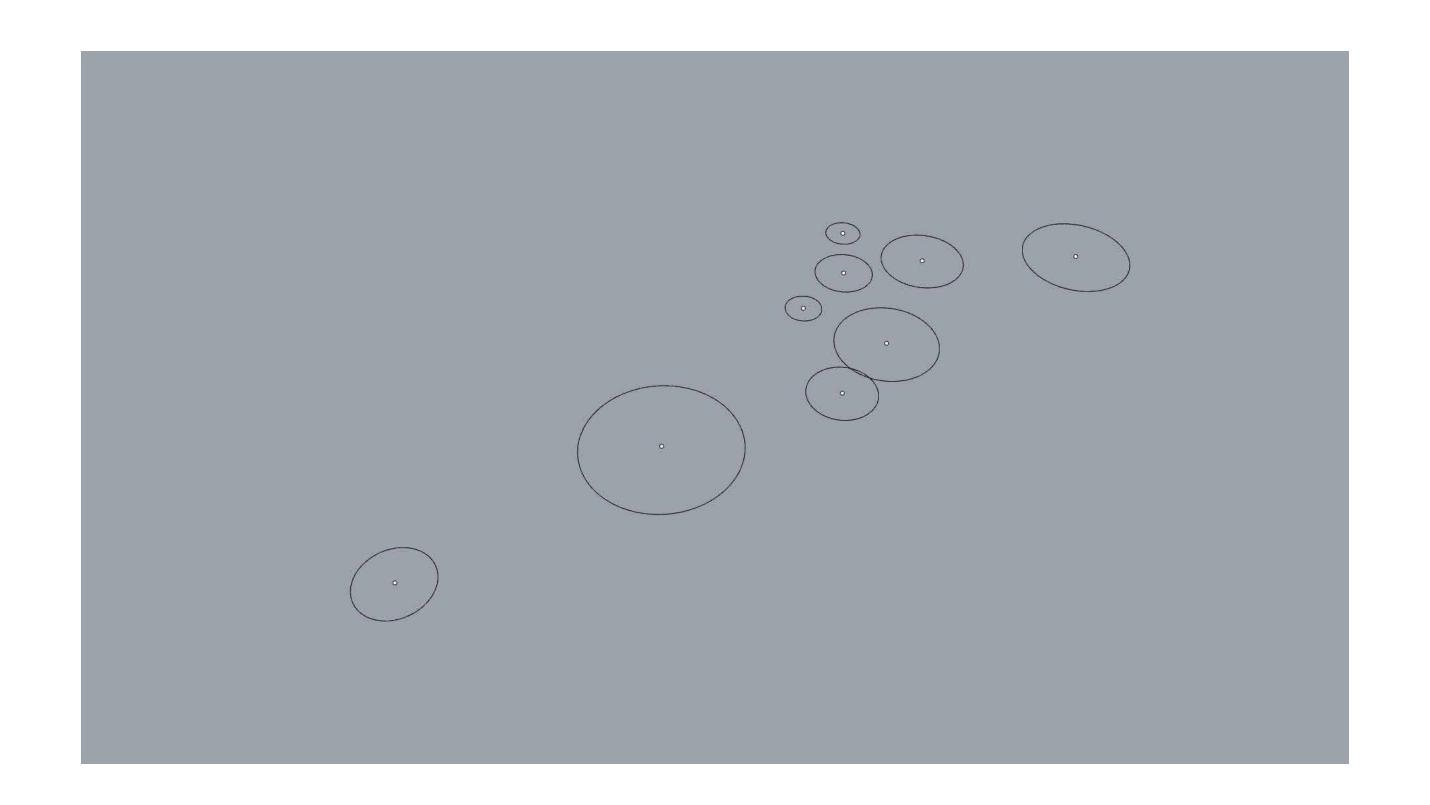
Based on Internal Risk Levels



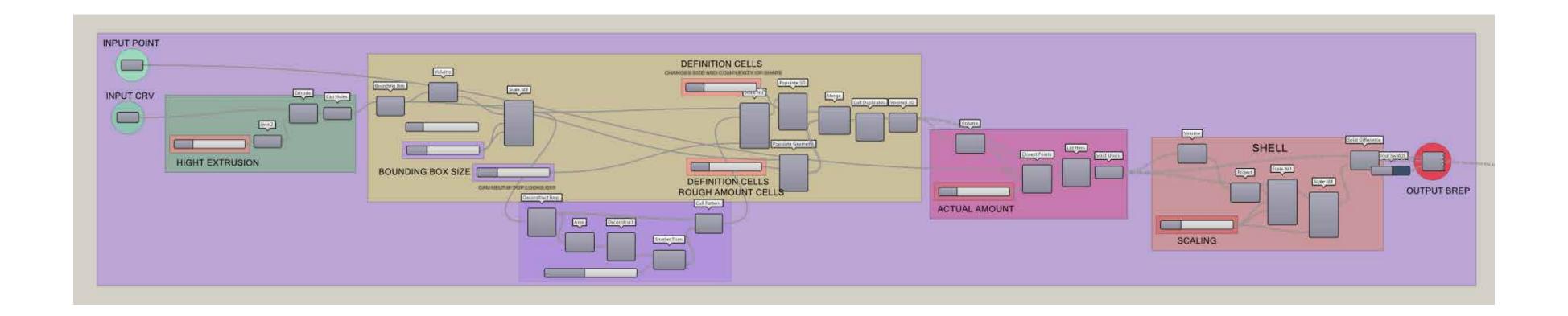
### GRASSHOPPER SCRIPT

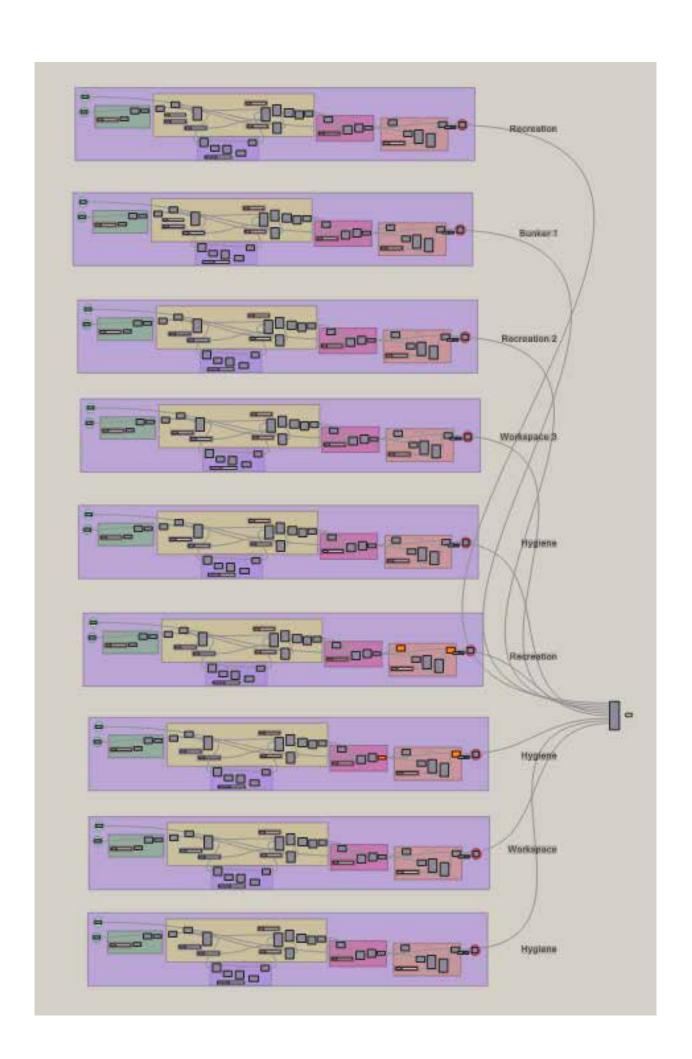
Obtaining positions and sizes according to the use and risk level

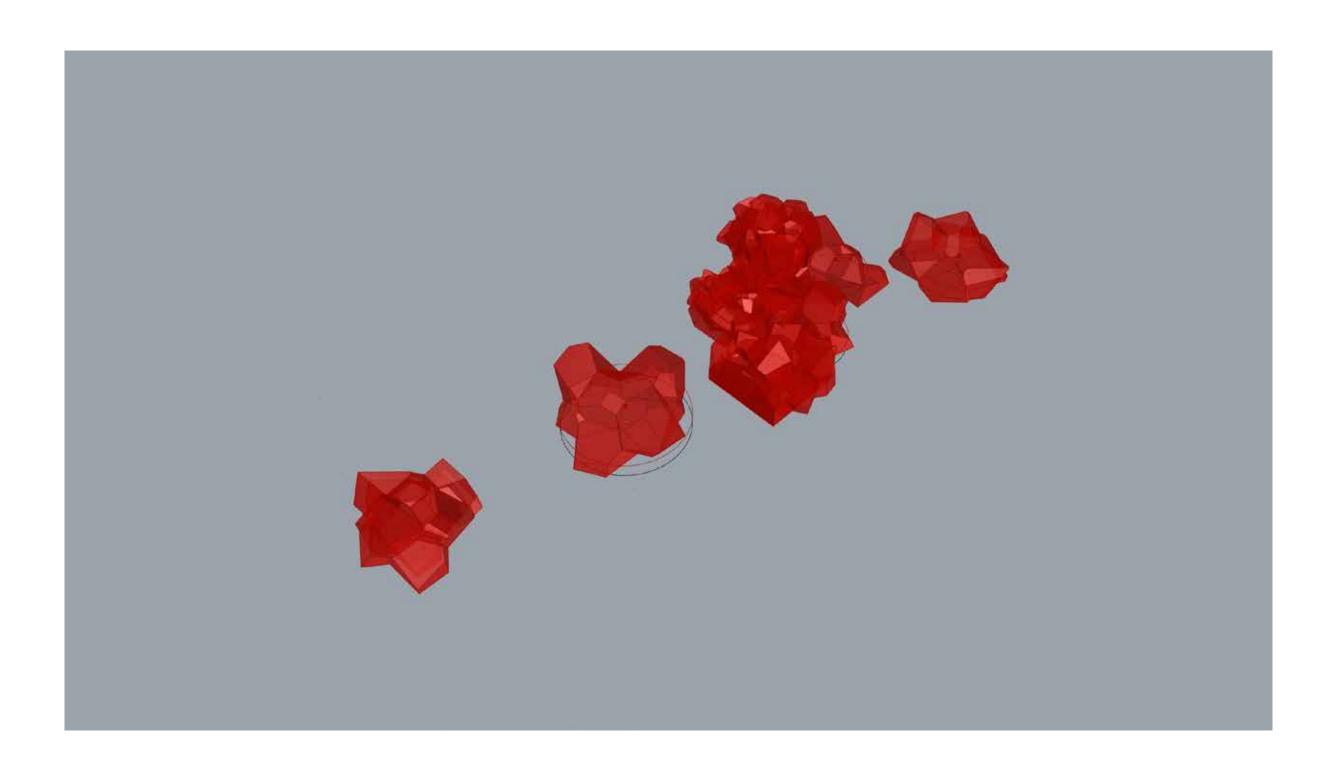




#### Generation of Voronoi geometry of the rooms



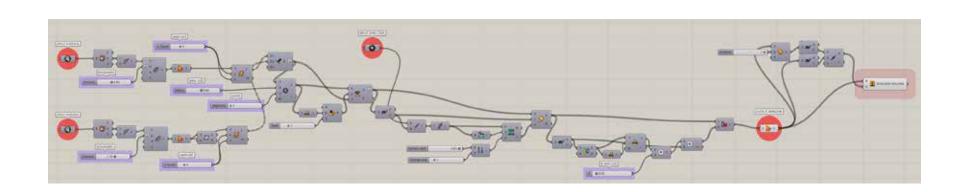


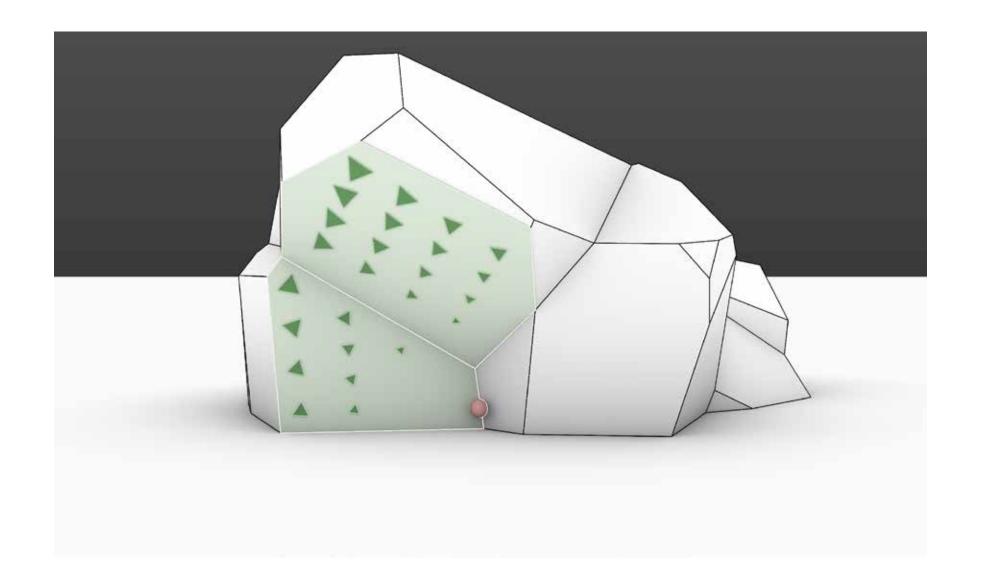


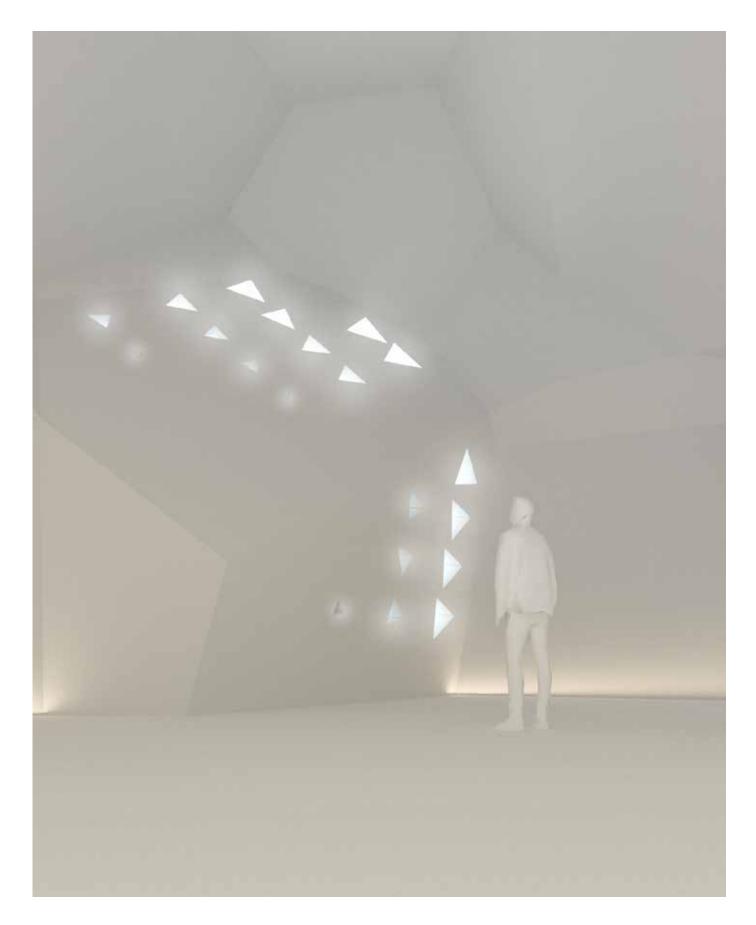
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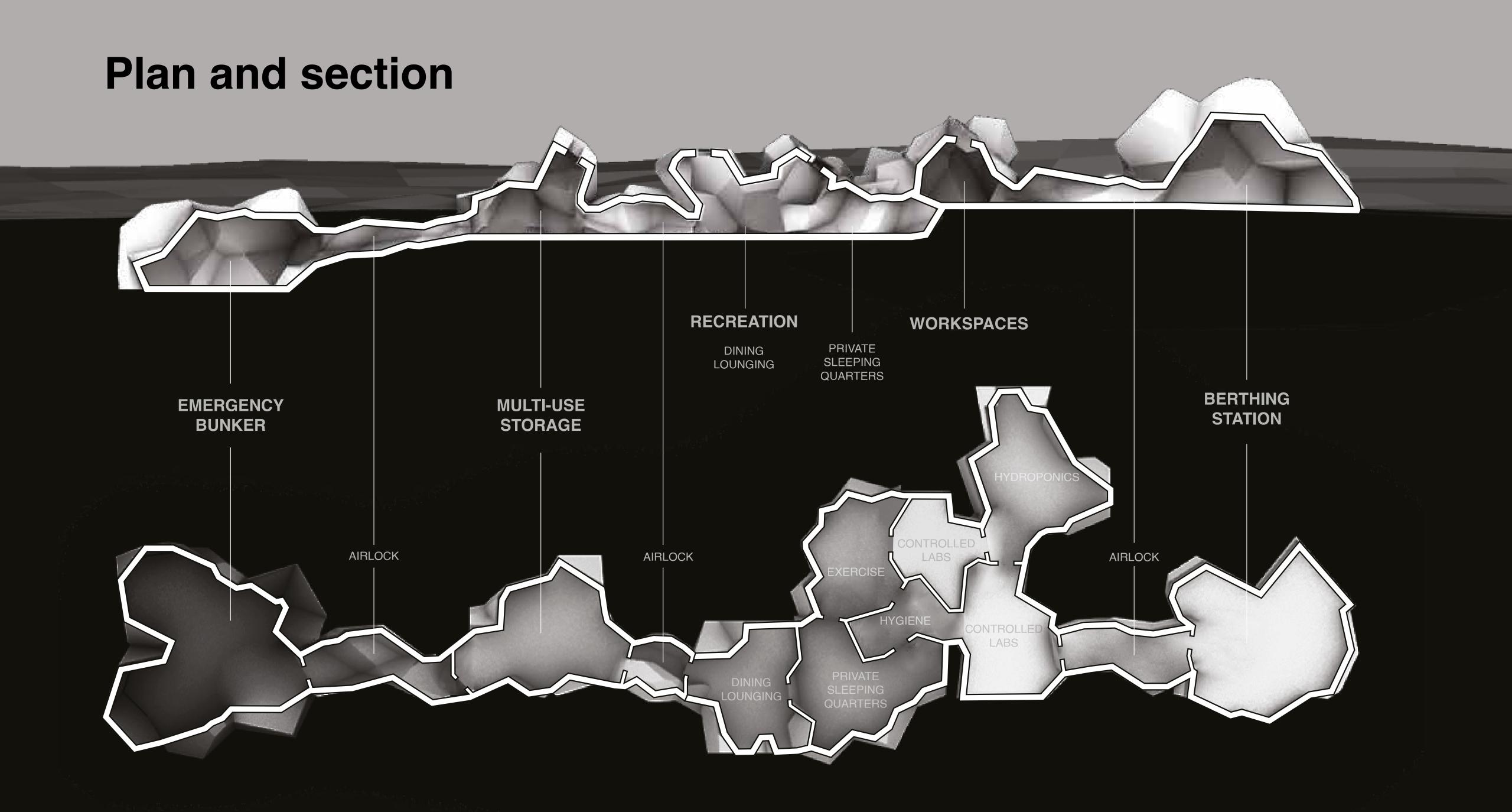


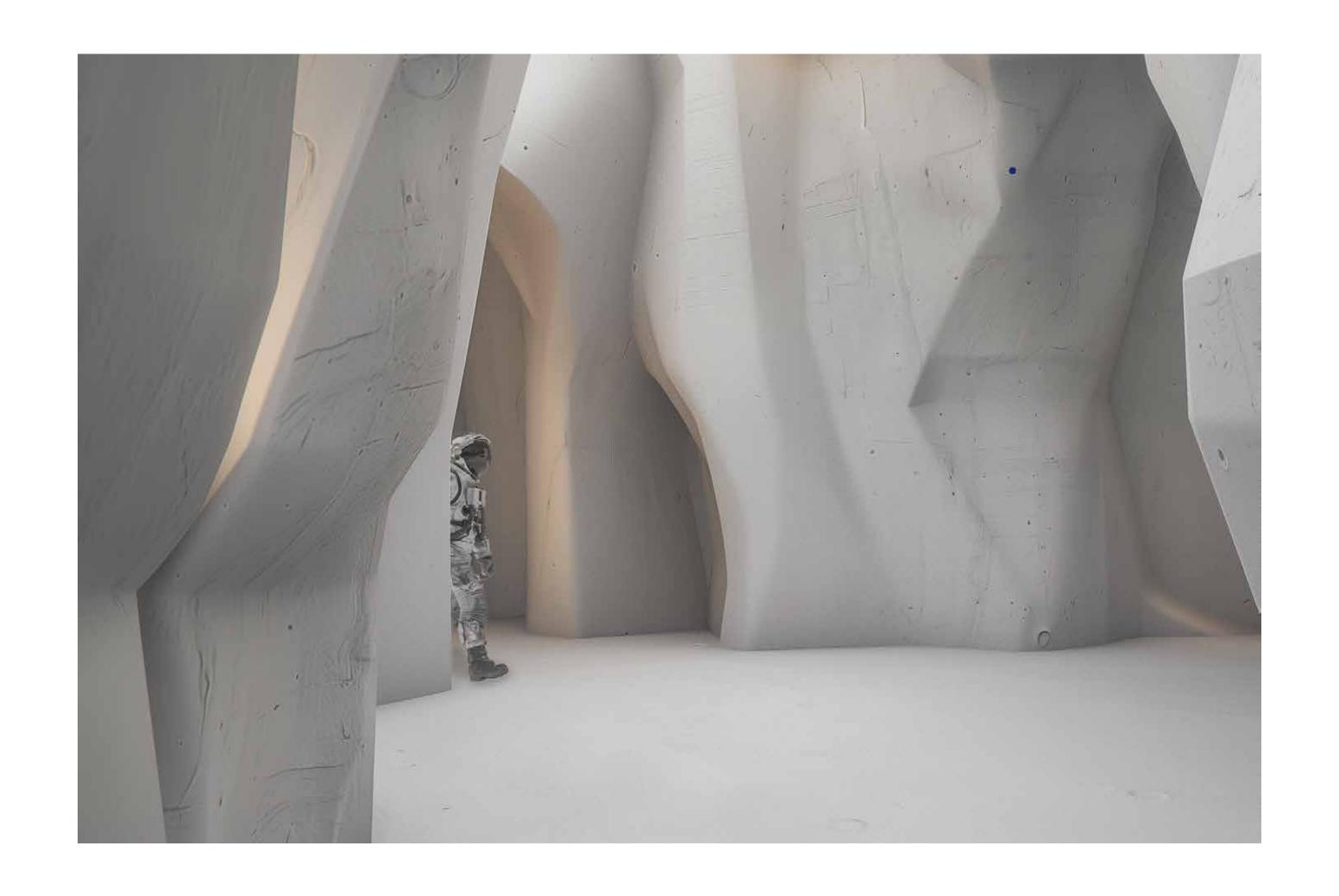
#### Generation of windows:

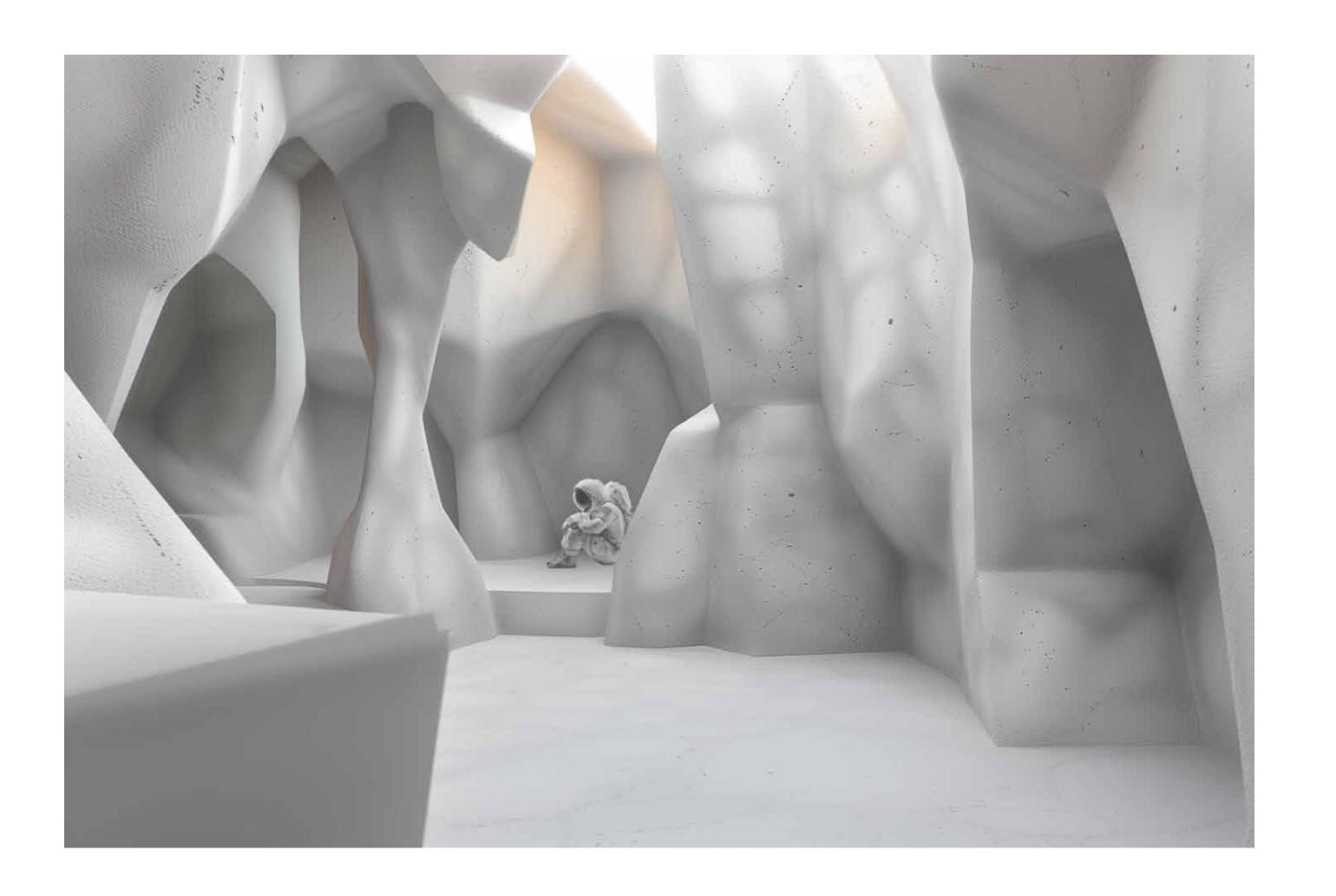












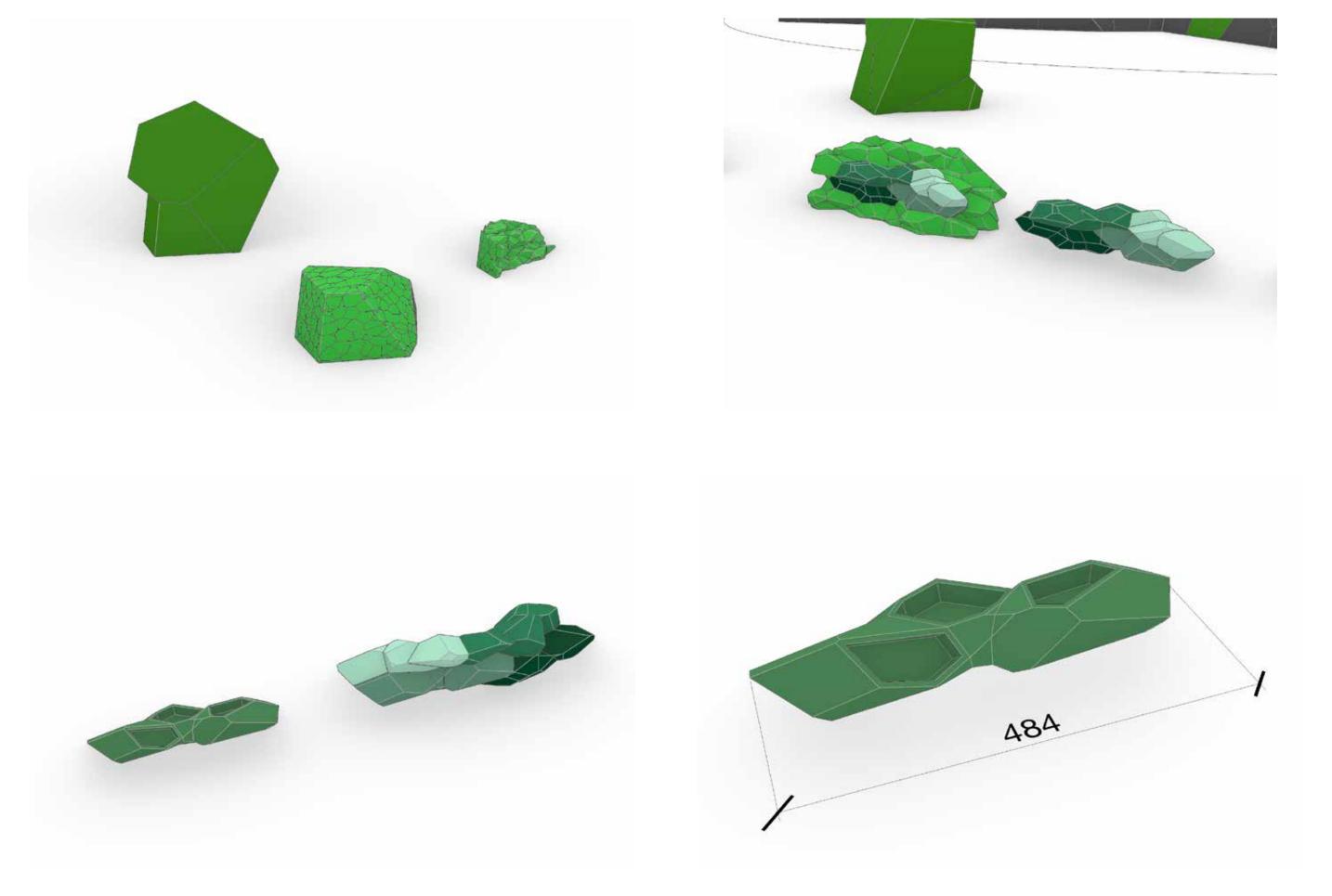




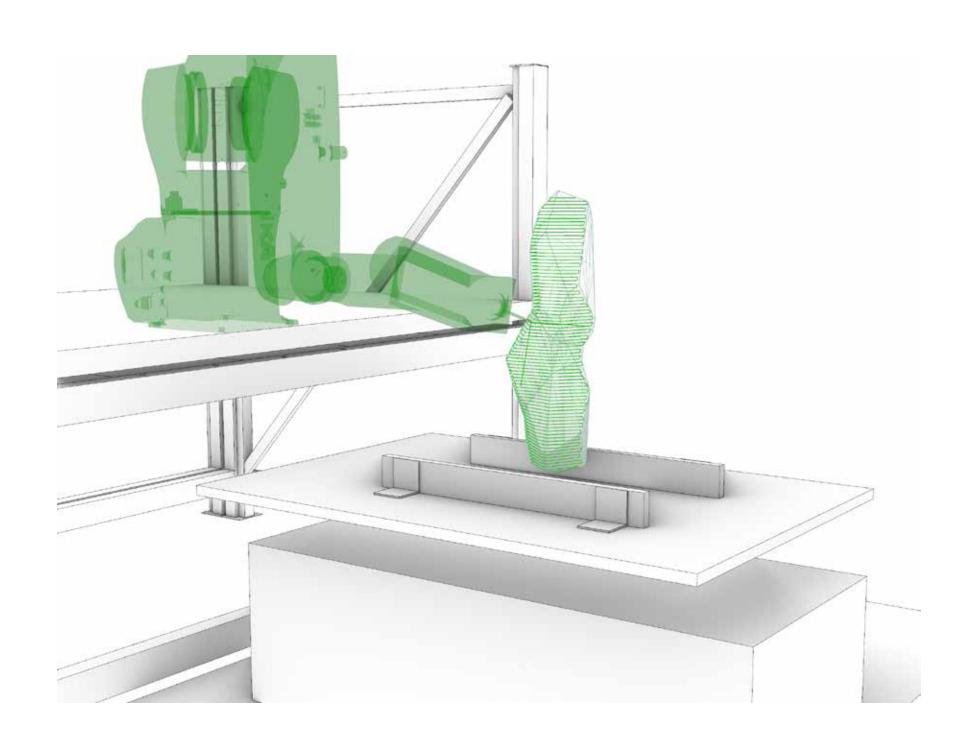


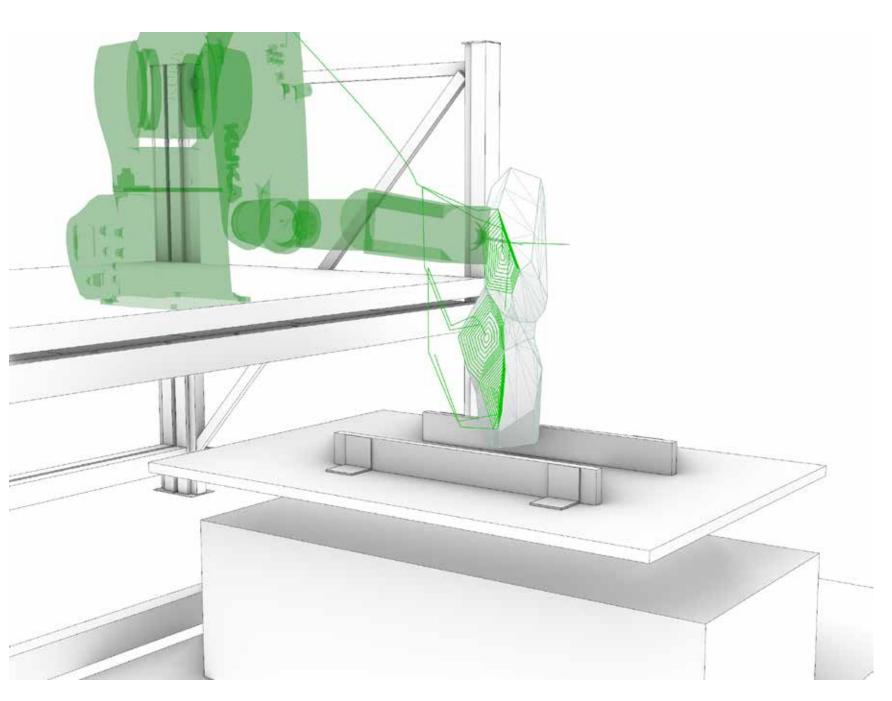
## 6\_PRODUCTION

#### Fragment extraction

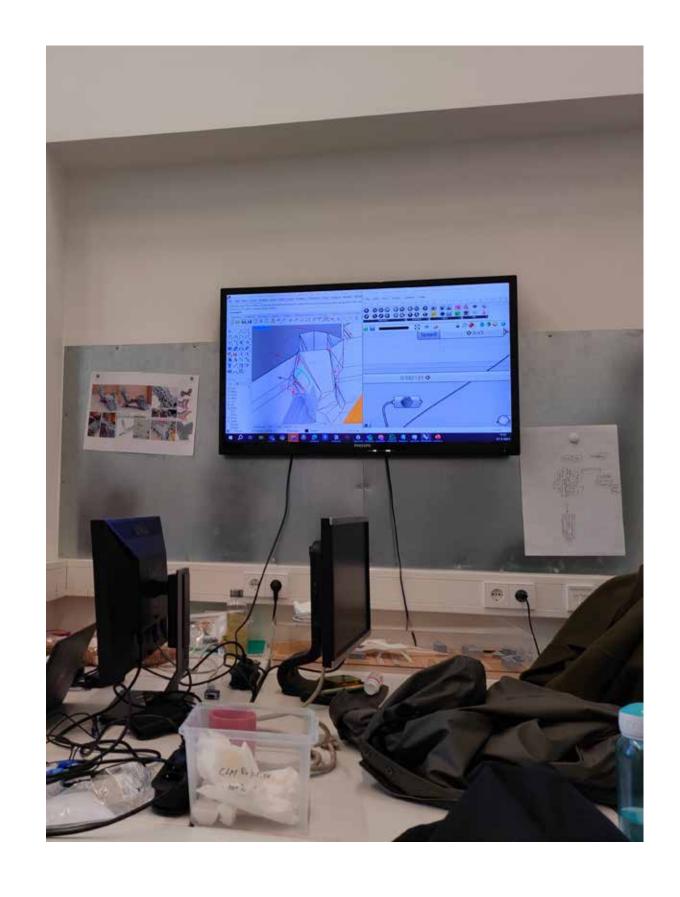


### Fragment milling preparation



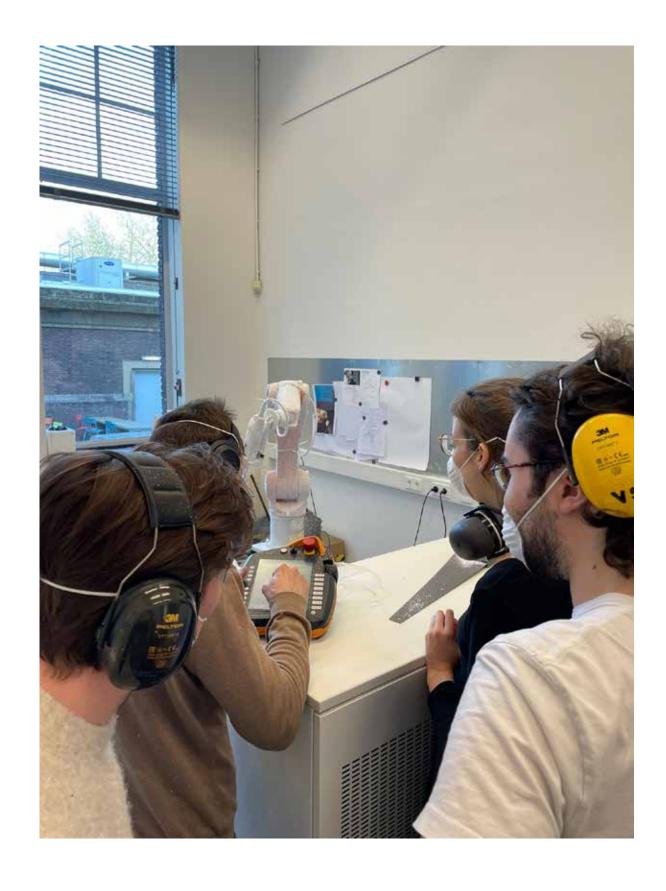


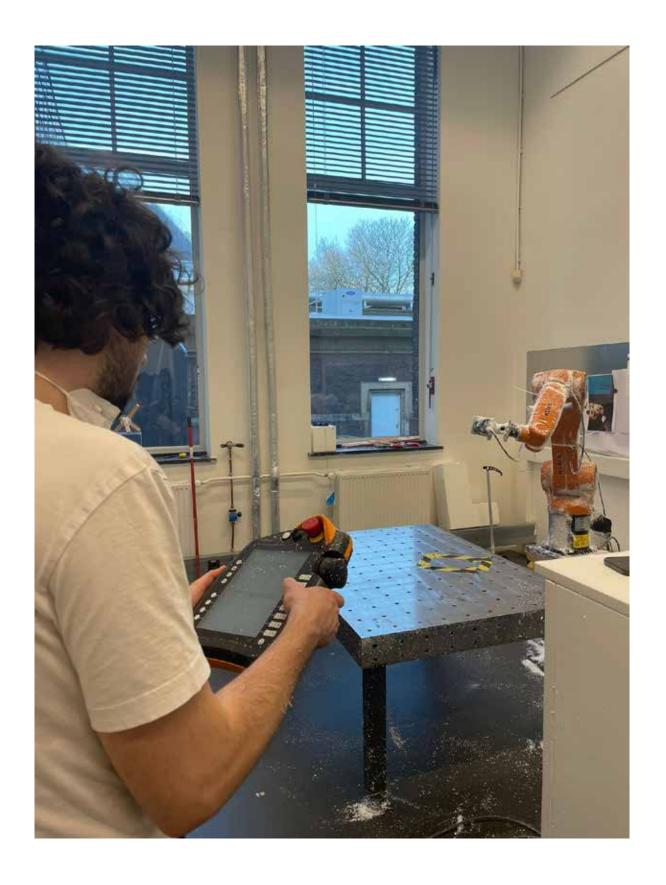
### Fragment production in the RB Lab:











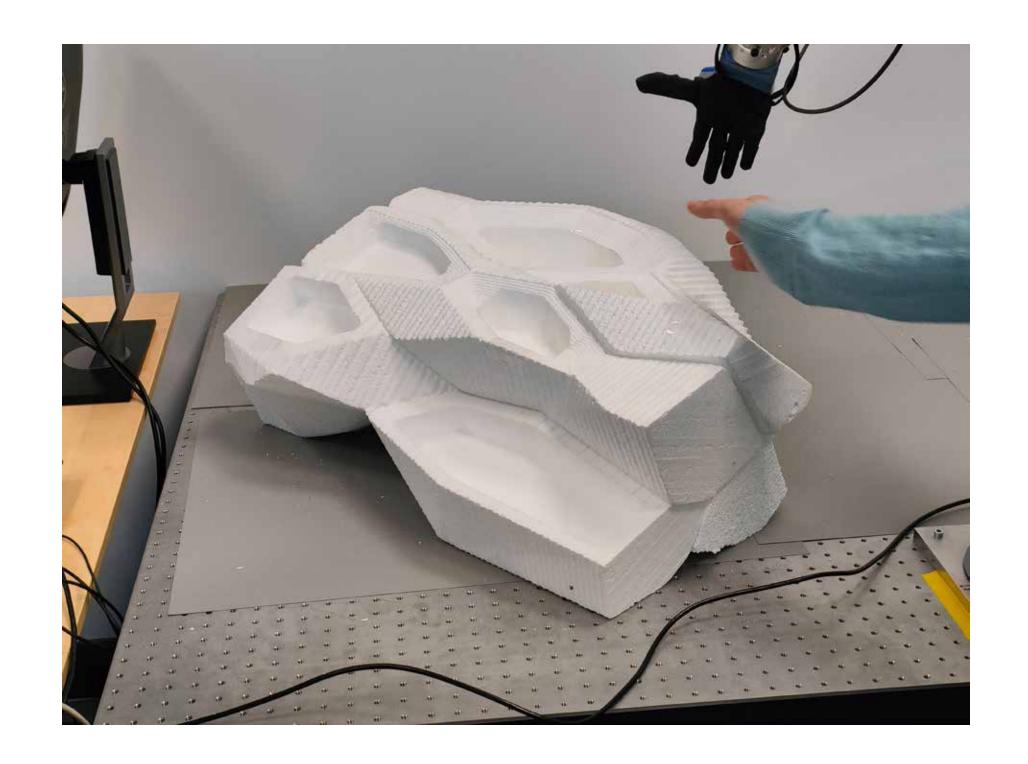


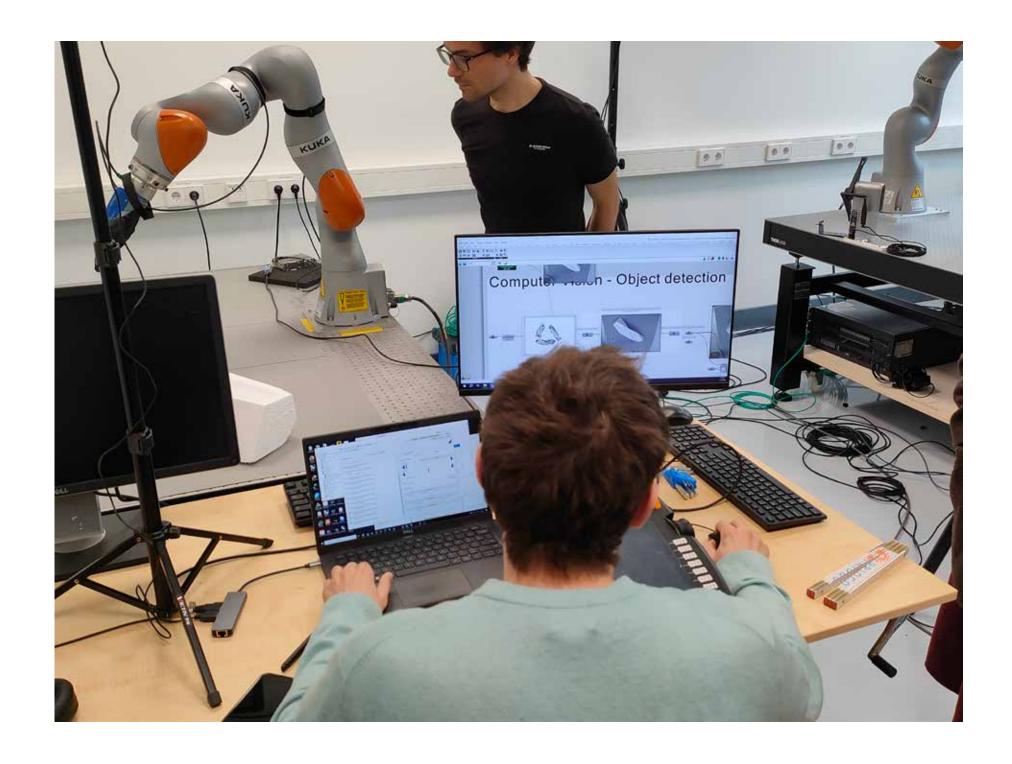
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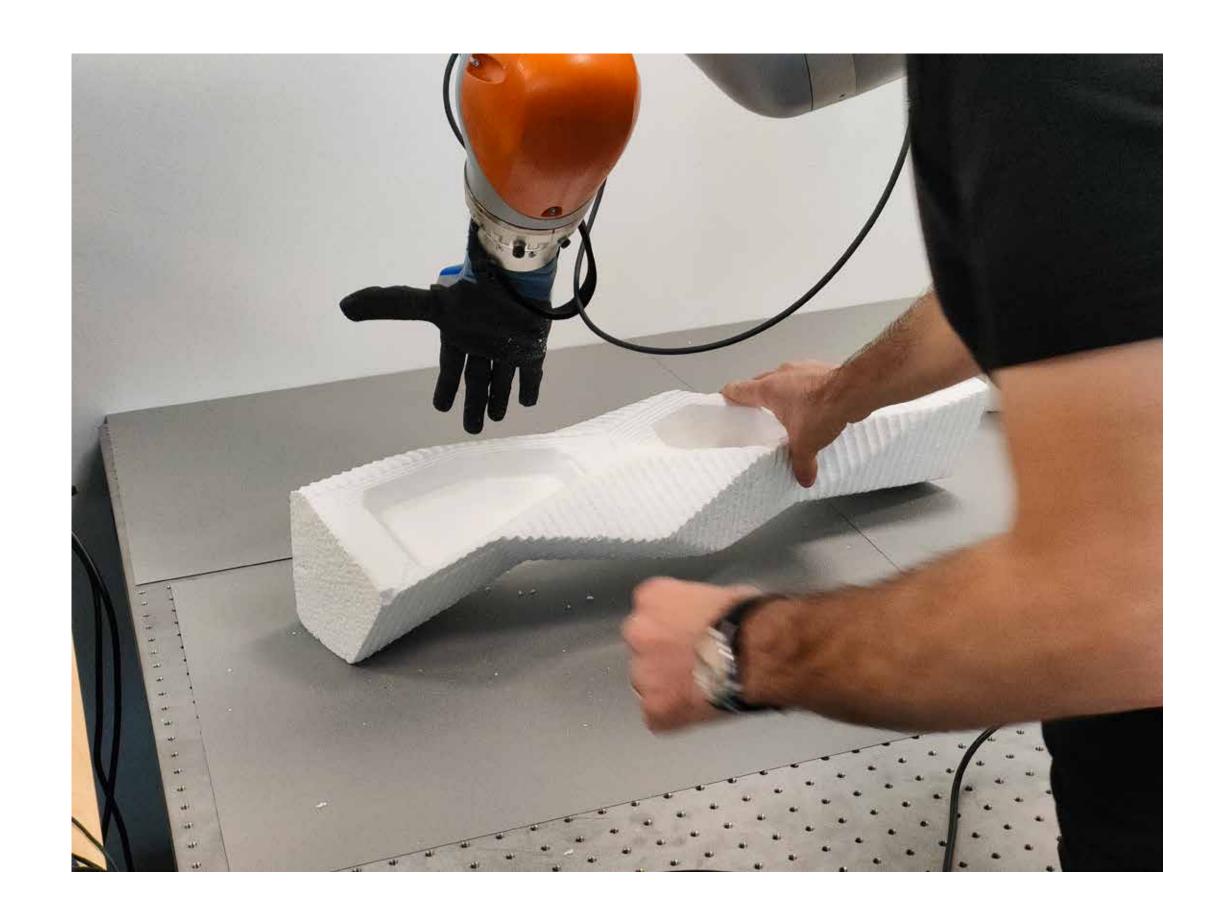


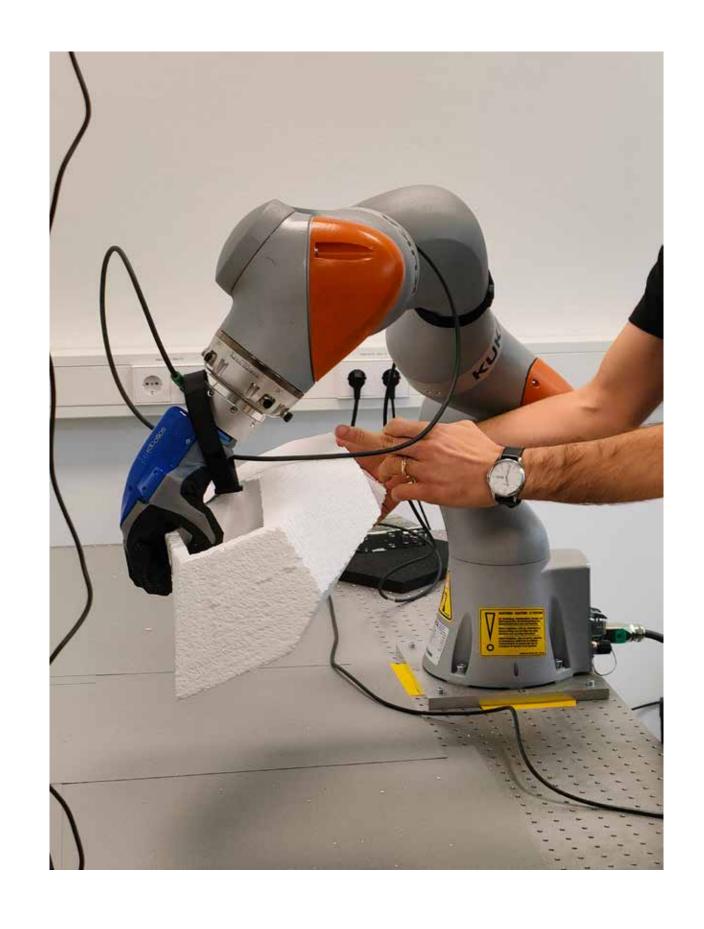
## 7\_ HUMAN-ROBOTIC INTERACTION

#### HRI demonstration in the CoR Lab:









# 8\_3D PRINTING

### 3D-printing of a fragment of the shelter:





